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EMC ANALYSIS OF THE PROPOSED AEROSAT VHF SUBSYSTEM WITH CURRENT INBAND COMMUNICATIONS SYSTEMS VOLUME I (USA AND CANADA)

IIT Research Institute
Under Contract to
DEPARTMENT OF DEFENSE
Electromagnetic Compatibility Analysis Center
Annapolis, Maryland 21402



May 1977

FINAL REPORT



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Prepared for

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION
Systems Research & Development Service
Washington, DC 20590

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PREFACE

The Electromagnetic Compatibility Analysis Center (ECAC) is a Department of Defense facility, established to provide advice and assistance on electromagnetic compatibility matters to the Secretary of Defense, the Joint Chiefs of Staff, the military departments and other DoD components. The Center, located at North Severn, Annapolis, Maryland 21402, is under executive control of the Assistant-Secretary of Defense for Communication, Command, Control, and Intelligence and the Chairman, Joint Chiefs of Staff, or their designees, who jointly provide policy guidance, assign projects, and establish priorities. ECAC functions under the direction of the Secretary of the Air Force and the management and technical direction of the Center are provided by military and civil service personnel. The technical operations function is provided through an Air Force sponsored contract with the IIT Research Institute (IITRI).

This report was prepared for the Systems Research and Development Service of the Federal Aviation Administration in accordance with Interagency Agreement DOT-FA70WAI-175, as part of AF Project 649E under Contract F-19628-76-C-0017, by the staff of the IIT Research Institute at the Department of Defense Electromagnetic Compatibility Analysis Center.

To the extent possible, all abbreviations and symbols used in this report are taken from American Standard Y10.19 (1967) "Units Used in Electrical Science and Electrical Engineering" issued by the USA Standards Institute.

Reviewed by:

J. M. DETERDING

Director of Contractor Operations

Approved by:

Project Engineer, IITRI

R. E. BEERY Colonel, USAF

Director

M. A. SKEATH

Special Projects

Deputy Director

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FEDERAL AVIATION ADMINISTRATION SYSTEMS RESEARCH AND DEVELOPMENT SERVICE SPECTRUM MANAGEMENT STAFF

STATEMENT OF MISSION

The mission of the Spectrum Management Staff is to assist the Department of State, Office of Telecommunications Policy, and the Federal Communications Commission in assuring the FAA's and the nation's aviation interests with sufficient protected electromagnetic telecommunications resources throughout the world to provide for the safe conduct of aeronautical flight by fostering effective and efficient use of a natural resource—the electromagnetic radio-frequency spectrum.

This objective is achieved through the following services:

- Planning and defending the acquisition and retention of sufficient radio-frequency spectrum to support the aeronautical interests of the nation, at home and abroad, and spectrum standardization for the world's aviation community.
- Providing research, analysis, engineering, and evaluation in the development of spectrum related policy, planning, standards, criteria, measurement equipment, and measurement techniques.
- Conducting electromagnetic compatibility analyses to determine intra/inter-system viability and design parameters, to assure certification of adequate spectrum to support system operational use and projected growth patterns, to defend the aeronautical services spectrum from encroachment by others, and to provide for the efficient use of the aeronautical spectrum.
- Developing automated frequency-selection computer programs/routines to provide frequency planning, frequency assignment, and spectrum analysis capabilities in the spectrum supporting the National Airspace System.
- Providing spectrum management consultation, assistance, and guidance to all aviation interests, users, and providers of equipment and services, both national and international.



EXECUTIVE SUMMARY

As a joint venture of the United States (US), Canada and the European Space Agency (ESA), an experimental aeronautical satellite (AEROSAT) system is presently being developed. AEROSAT will provide evaluation of communications and position-fixing techniques for aircraft flying transoceanic routes.

The Federal Aviation Administration (FAA) is directing US participation in the program. FAA has requested that the DoD Electromagnetic Compatibility Analysis Center (ECAC) determine means by which the AEROSAT satellite-to-aircraft VHF link can operate compatibly with the inband international air-to-ground communications systems.

Operational conditions were evaluated to determine possible interactions between AEROSAT and the VHF user community. Usage in the VHF frequency band (125.4-132.0 MHz) was ascertained and representative VHF equipment electrical parameters were determined. AEROSAT EMC equipment specifications and operational constraints were established to preclude interference to the VHF-user community. The FAA-proposed frequency plan, whereby AEROSAT frequencies would be assigned midway between the 50-kHz channels presently used in the 125.4-126.0 and 131.4-132.0 MHz frequency bands, was shown to be feasible.

Only the 1975 US and Canadian environment was addressed as a direct application of the generalized analysis. Further analysis is required for the Atlantic area. The results show that, if AEROSAT avionics and satellite receivers are developed to meet the recommended adjacent-channel rejection requirements, no restrictions need be placed on AEROSAT VHF frequency assignments for operation in North America.

Co-channel operation of Aerosat and other VHF-users was not considered in this analysis.

FAA-RD-77-56, I

TABLE OF CONTENTS

Subsection	Page
SECTION 1	
INTRODUCTION	
BACKGROUND	1
OBJECTIVES	2
APPROACH	2
SECTION 2	
SYSTEM DESCRIPTION	
AEROSAT SYSTEM	5
AEROSAT VHF SUBSYSTEM	6
General Description	6
Satellite Description	8
Avionics Description	10
TERRESTRIAL SYSTEM	13
SECTION 3	
ANALYSIS	
POTENTIAL INTERFERENCE SITUATIONS	19
ANALYSIS PARAMETERS	23
General	23
Antennas and Transmission Lines	23
Propagation	27
Ionospheric Effects	27
Multipath	29
Receiver Characteristics	38
CALCULATIONS	46
The Necessity of Interleaving	46
Interference to the AEROSAT System by the Terrestrial	
System	48

TABLE OF CONTENTS (Continued)

Subsection	Page
SECTION 3 (Continued)	
Interference to the Terrestrial System by the AEROSAT	
System	54
The Sufficiency of Interleaving	59
Interference in the US and Canada	60
Interference to the AEROSAT Satellite	61
SECTION 4	
RESULTS	
RESULTS	67
LIST OF ILLUSTRATIONS	
Figure	
1 ATLANTIC SATELLITE COVERAGE AREAS	7
2 EFFECTIVE SATELLITE TRANSCEIVER	9
3 BLOCK DIAGRAM OF AEROSAT AVIONICS TRANSCEIVERS	11
4 TERRESTRIAL SYSTEM POTENTIAL INTERFERENCE TO THE	
SATELLITE RECEIVER	19
5 TERRESTRIAL SYSTEM GROUND TRANSMITTER POTENTIAL	
INTERFERENCE TO THE AEROSAT AVIONICS RECEIVER	20
6 TERRESTRIAL SYSTEM AIRBORNE TRANSMITTER POTENTIAL	
INTERFERENCE TO THE AEROSAT AVIONICS RECEIVER	20
7 AEROSAT SATELLITE TRANSMITTER POTENTIAL INTER-	
FERENCE TO THE TERRESTRIAL SYSTEM GROUND RECEIVER	21
8 AEROSAT SATELLITE TRANSMITTER POTENTIAL INTER-	
FERENCE TO THE TERRESTRIAL SYSTEM AIRBORNE	
RECEIVER	21
9 AEROSAT AIRBORNE TRANSMITTER POTENTIAL INTER-	
FERENCE TO THE TERRESTRIAL SYSTEM GROUND	
RECEIVER	22

FAA-RD-77-56, I

TABLE OF CONTENTS (Continued)

Page

LIST OF ILLUSTRATIONS (Continued)

Figure		Page
10	AEROSAT AIRBORNE TRANSMITTER POTENTIAL	
	INTERFERENCE TO THE TERRESTRIAL SYSTEM	
	AIRBORNE RECEIVER	22
11	BOEING 747 SATCOM ANTENNA VOLTAGE PATTERN FOR	
	LINEAR POLARIZATION (PITCH PATTERN)	24
12	BOEING 747 SATCOM ANTENNA VOLTAGE PATTERN FOR	
	LINEAR POLARIZATION (ROLL PATTERN)	25
13	MAP WITH INVARIANT GEOMAGNETIC COORDINATES	28
14	SCINTILLATION LEVELS	30
15 •	MULTIPATH DIAGRAM	31
16	REFLECTION COEFFICIENT OF SEA WATER AT 130 \mbox{MHz} .	33
17	AIR-GROUND PROPAGATION IN THE TERRESTRIAL SYSTEM	35
18	PROPAGATION BETWEEN AN AEROSAT AIRPLANE AND A	
	TERRESTRIAL SYSTEM GROUND STATION	36
19	PROPAGATION BETWEEN AN AEROSAT AIRPLANE AND A	
	TERRESTRIAL SYSTEM AIRPLANE	37
20	PHASE LOCKED LOOP DETECTOR	39
21	ARTICULATION SCORE AS A FUNCTION OF ARTICULATION	
	INDEX, FOR FM VOICE OR AM VOICE	45
22	DEGRADATION THRESHOLD CURVES FOR THE CASE OF	
	AEROSAT INTERFERENCE 25 kHz OFF TUNE TO	
	TERRESTRIAL SYSTEM RECEIVERS	47
	LIST OF TABLES	
Table		
1	VHF SCINTILLATION LEVELS (APPROXIMATE)	29

TABLE OF CONTENTS (Continued)

LIST OF APPENDIXES

Appendix		Page
A	TRANSMITTERS IN THE US AND CANADA IN THE	
	131.4 to 132.0 MHz BAND	69
В	TRANSMITTERS IN THE US AND CANADA IN THE	
	125.4 to 126.0 MHz BAND	81
С	COVERAGE AREAS OF EXTENDED RANGE VHF STATIONS	
	IN THE US, CANADA, AND THE NORTH ATLANTIC	
	REGION	95
REFERENCE	S	103

SECTION 1

INTRODUCTION

BACKGROUND

In 1974 a Memorandum of Understanding (MOU) was signed by the United States, Canada, and the European Space Agency (formerly the European Space Research Organization). The MOU provides for the implementation of an Aeronautical Satellite System (AEROSAT) to experiment with and develop an operational capability for improving oceanic air traffic control. The system will be initially deployed over the Atlantic Ocean but expansion to the Pacific is planned. AEROSAT will provide two-way voice and data communications, via satellite, between ground locations and aircraft flying over the oceanic area.

Part of the AEROSAT system will include a satellite-aircraft link operating in the VHF band. This band is presently being used internationally for air-to-ground communications by commercial and private aircraft (hereafter referred to as the terrestrial system in this report). It is desired that no changes be made to the terrestrial systems for compatible operation of the two systems.

The Federal Aviation Administration (FAA), which is directing the U.S. participation in the AEROSAT program, tasked the Electromagnetic Compatibility Analysis Center (ECAC) to conduct an electromagnetic compatibility (EMC) analysis of the AEROSAT system. In addition to determining the EMC between the AEROSAT VHF subsystem and

Annex I and Annex II to the Memorandum of Understanding on a Joint Programme of Experimentation and Evaluation Using an Aeronautical Satellite Capability Between the US Dept. of Transportation, Federal Aviation Administration (FAA), the European Space Research Organization (ESRO), and the Gov. of Canada, August 2, 1974.

systems operating in the same band, ECAC was asked to determine the extent of the interference-free areas within which AEROSAT-equipped aircraft could operate.

OBJECTIVES

The objectives of this task were to:

- 1. Determine the compatibility of **EROSAT-VHF subsystem with the present terrestrial systems, and if required, to develop methods that will provide for compatible operation.
- 2. Recommend guidelines, utilizing methods developed, that will provide for compatible operation between the AEROSAT-VHF subsystem and its environment in the US and Canada.

APPROACH

The analysis was performed in five (5) steps.

- 1. Technical data and operational procedures were collected on the systems being considered from ECAC files and other sources. Data on the AEROSAT system was provided to ECAC by the FAA, while data on the terrestrial systems was provided to ECAC by both the FAA and Aeronautical Radio Incorporated (ARINC).
- 2. Potential interference paths were determined. Satellite-to-aircraft or aircraft-to-satellite communications in the AEROSAT system may be sources of interference to, or experience interference from air-to-ground or ground-to-air communications in the terrestrial systems.
- 3. The parameters affecting the analysis were studied. These parameters were:
 - a. scintillation of transionospheric signals,
 - b. multipath propagation,

- c. the effects of antenna patterns on propagation and,
- d. receiver performance in the presence of interference.
- 4. The results of the parameter analysis were applied to the potential interference paths to determine if the proposed AEROSAT system can operate compatibly with the terrestrial systems. Simple procedures were developed for AEROSAT implementation that will not require any changes to the terrestrial system.
- $\,$ 5. The procedures developed in step 4 were applied to the US and Canadian systems.

SECTION 2

SYSTEM DESCRIPTION

AEROSAT SYSTEM

The AEROSAT system will provide two-way voice and data communications, via satellite, between ground locations and aircraft flying over the oceanic area. A requirement is that AEROSAT operate in all oceanic areas not covered by the terrestrial system and it is desired that AEROSAT operate in all areas, both oceanic and over land.

Three frequency bands will be utilized by the AEROSAT system.

- 1. 125.4 to 126.0 MHz; 131.4 to 132.0 MHz (VHF). Satellite-to-aircraft links will use the 125.4 to 126.0 MHz bend. Aircraft-to-satellite links will use the 131.4 to 132.0 MHz band.
- 2. 1543.5 to 1660 MHz. Satellite-to-aircraft links will use the 1543.5 to 1578.5 MHz band. Aircraft-to-satellite links will use the 1622.5 to 1636.5 MHz and 1645 to 1660 MHz band.
- 3. 5000 to 5250 MHz. Ground-to-satellite links will use the 5000 to 5125 MHz band. Satellite-to-ground links will use the 5125 to 5250 MHz band.

Two geostationary satellites will be located between 15 and 40 degrees West longitude for the Atlantic system. Initially, three ground terminals will be built, one in Europe, one in Canada, and one in the United States. Avionics installations will be made on selected commercial aircraft: Boeing 747, Lockheed L-1011, and Douglas DC-10. The VHF avionics will be installed on the 747 aircraft only.

The two-satellite system has the potential for providing aircraft position fixing by comparing the two satellite signals. The program

will experiment with aircraft position fixing and will develop operational standards and procedures.

The AEROSAT system development will approximately follow this schedule.

- 1976: Design of ground terminals, avionics, and satellites.
- 1977: Begin construction of ground terminals, first avionics units, and satellites. Complete construction and begin testing of first avionics units.
- 1978: Complete construction and begin testing of ground terminals and satellites. Begin construction of the remaining avionics units. Begin system test.
- 1979: Launch first satellite. Additional system tests. Launch second satellite.

The system will become fully operational in the early 1980's. FIGURE 1 shows the areas to be initially served by AEROSAT. System performance requirements must be met in all areas where the satellite elevation exceeds 10° .

AEROSAT VHF SUBSYSTEM

General Description

The AEROSAT VHF subsystem is designed to handle analog or digital signals. The analog signals will be FM voice with a maximum peak deviation of 1.5 kHz. The frequency range will be limited to voice frequencies (300 to 3,000 Hz) and a pre-emphasis, de-emphasis of 6 dB per octave will be used, to reduce the noise level.

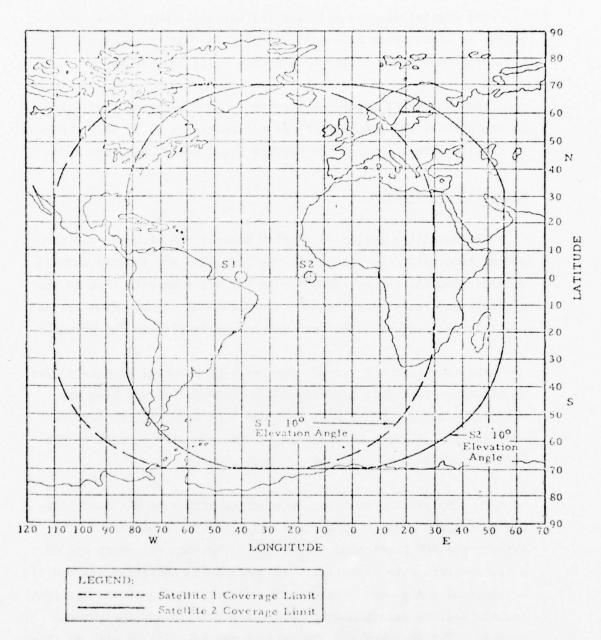


FIGURE 1. ATLANTIC SATELLITE COVERAGE AREAS.

The digital signals will use Differential Phase Shift Keying (DPSK) to communicate data at 1,200 or 2,400 bits per second. The bit error rate (BER) is to be less than 1 bit per 100,000.

The US AEROSAT Candidate Frequency Plan² proposes that carriers in the AEROSAT VHF subsystem be at odd multiples of 25 kHz. However, no final determination of the carrier frequencies, has been made.

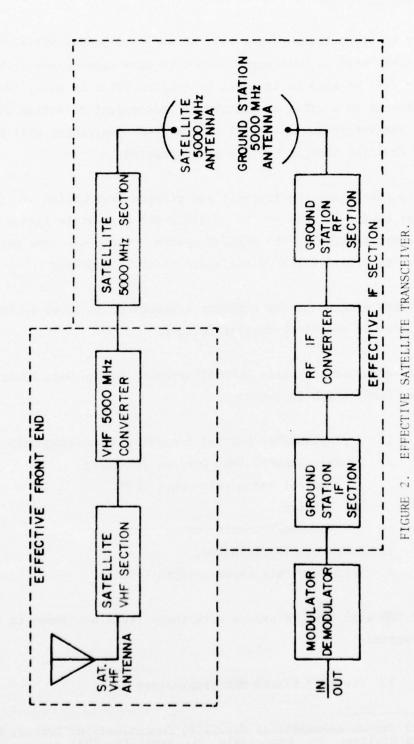
Satellite Description

The satellite serves as a relay between the aircraft and the ground terminals. It translates the 5000-5125 MHz forward channel signals from the ground terminals to VHF and transmits them to the aircraft and translates the VHF return channel signals from the aircraft to the 5125-5250 MHz band and transmits them to the ground.

The satellite VHF transmitters will be powerful enough to give an Effective Isotropic Radiated Power (EIRP) of 25 dBW from the antenna. Each satellite will be capable of handling two forward channels and four return channels at once and will be able to combine the two forward channels for double power.

To simplify the analysis, the entire 5000-5250 MHz band link can be thought of as an IF section and modulator/demodulator. In effect, the VHF portion of the satellite becomes the front end of a transceiver. The selectivity and modulation characteristics will be those of the ground station. The sensitivity and noise specifications will be determined by both the satellite and the ground station. The RF power and the antenna characteristics will be those of the satellite. This "effective transceiver" will be referred to as the "satellite transceiver." FIGURE 2 is a block diagram depicting the "satellite transceiver."

²The Federal Aviation Administration, Candidate Frequency Plan, AEROSAT, US Draft, February 1975.



9

Avionics Description

The airborne transceivers will be modified Bendix RTA-42A³ units. They will be half-duplex voice or data transceivers; that is, they will be able to transmit or receive voice or data, switching from one mode to another as required. Independent selection of transmit and receive frequencies will be provided. Switching will be controlled from the cockpit or by a mini computer.

Note that since the transmit and receive frequencies are in different bands, it will not be possible for a pilot to listen to other pilots talking on the AEROSAT system. Therefore some method of informing pilots which channels are clear is necessary.

A block diagram of the airborne transceiver is shown in FIGURE 3. A discussion of the block diagram follows.

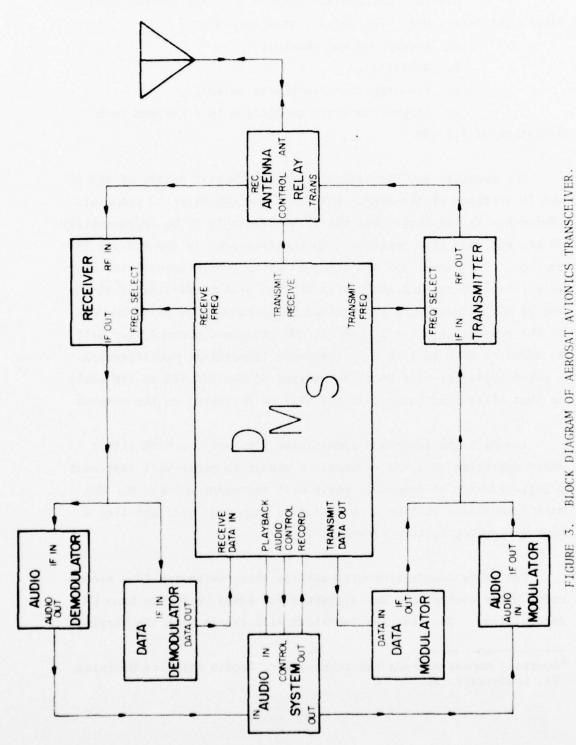
The center of the whole AEROSAT avionics is the Data Management Subsystem (DMS) which includes:

- 1. A programmable Digital Processor (Mini-computer),
- Cockpit Input/Output Devices including;
 - a. Control and Display unit (CDU)
 - b. Printer
 - c. Receive/Transmit Control unit,
- 3. Cabin Display/Keyboard,
- 4. Voice and Data Record/Playback units.

The DMS will also interface with these items not shown in the block diagram:

1. The 1543.5-1660 MHz transceiver

³Proposal for an Aeronautical Satellite Communications System, Bendix Avionics Division, Ft. Lauderdale, FL, April 15, 1973.



11

- Aircraft peripherals such as a Greenwich Mean Time clock, altimeter, etc. The audio system includes:
 - a. Microphone and speaker,
 - b. Amplifier,
 - c. Pre-emphasis/de-emphasis networks,
- $\mbox{d. Clipper to limit modulation to a maximum peak} \\ \mbox{deviation of 1.5 kHz.}$

The receiver and the transmitter will consist mainly of the RF and IF sections of the Bendix RTA-42A. The modification proposals (Reference 3) indicates that the IF bandwidth is to be approximately 20 kHz and that this represents no modifications to the RTA-42A IF section. The manual for the RTA-42A⁴ gives two IF selectivity characteristics, broad and sharp with minimum 6-dB bandwidths of 30 kHz and 18 kHz respectively and maximum 60-dB bandwidths of 50 kHz and 39 kHz respectively. A 118 to 136 MHz broadband preamplifier will be added to give an 1100 K system noise temperature specification. A power amplifier will boost the output of the RTA-42A to 150 watts so that after line losses, 19 dBW will be delivered to the antenna.

The data modulator and demodulator will utilize DPSK with a phase deviation of \pm $\pi/2$ radians. A change in phase will represent a logic 1 while no change in phase will represent a logic 0. The data demodulator will be a phase-locked loop that will act like a high Q tracking filter (Reference 3).

The voice demodulator will utilize the combination of a secondorder phase locked loop and a limiter for adaptive narrow band FM demodulation. That is, the bandwidth will change with the varying

⁴Overhaul Manual RTA-42A VHF Transceiver, Bendix Avionics Division, Ft. Lauderdale, FL.

signal-to-noise ratio (S/N) becoming broader when the S/N is high for greater fidelity and becoming narrower when the S/N is low to reject as much noise as possible.

TERRESTRIAL SYSTEM

The VHF band to be used by AEROSAT is presently being used for air-to-ground communications by commercial and private aircraft. The 125.4 to 126.0 MHz band is being used for Air Traffic Control (ATC) while the 131.4 to 132.0 MHz band is being used for Operational Control (navigation and information). The operational control band is also used by Extended Range VHF stations. These stations, some of which are in the 131.4 to 132.0 MHz band, provide for communication with aircraft out to 400 miles over the ocean. The Extended Range VHF stations are of special importance to this analysis because it is the oceanic areas not covered by Extended Range VHF stations, that AEROSAT must cover. No Extended Range VHF stations are in the 125.4 to 126.0 MHz band.

The terrestrial system uses push-to-talk AM communications where a single channel is used (in a particular area) and everyone in the air and on the ground that is tuned to that frequency is listening to the one person who has switched to transmit and is talking. When two people try to talk at the same time, they interfere with each other.

The channels are 50-kHz wide, centered at multiples of 50 kHz. However by late 1977, many areas will begin converting to 25-kHz wide channels, centered at multiples of 25 kHz. ICAO Annex 10^5 provides the following channel-spacing information for the 117.975-136 MHz band.

⁵International Standards and Recommended Practices: Aeronautical Telecommunications Annex 10 to the Convention on International Civil Aviation, International Civil Aviation Organization, Quebec, Canada, July 1972.

4.1.2.2. After 1 July 1976 and until 1 January 1977, the minimum separation between assignable frequencies in the International Aeronautical Mobile Service shall be 25 kHz in those areas where such channel spacing has been introduced by regional agreement, and the minimum separation between assignable frequencies shall be 50 kHz in all other areas. After 1 January 1977, the minimum separation between assignable frequencies in the International Aeronautical Mobile Service shall be 25 kHz.

NOTE 1 - The intent of this paragraph is to allow those aircraft engaged in operations in areas where 25-kHz channel spacing will not be introduced prior to 1 January 1977 an additional period of time in which to effect the required airborne equipment modifications necessary to operate in a 25-kHz channel spacing environment.

NOTE 2 - It is recognized that, after 1 July 1976 and even after 1 January 1977 in some regions or areas, 100-kHz or 50-kHz channel spacing may provide an adequate number of frequencies suitably related to international and national air services and that equipment designed specifically for 100-kHz or 50-kHz channel spacing will remain adequate for services operating within such regions or areas.

Ground receivers operating within the 117.975-136 MHz band will be required to reject adjacent channels in accordance with the following (Reference 5).

4.6.2.3. Adjacent channel rejection. The receiving system shall ensure an effective rejection of 60 dB or more at the next assignable channel.

NOTE - The next assignable frequency will normally be plus or minus 50 kHz. Where this channel spacing will not suffice, the next assignable frequency will be plus or minus 25 kHz implemented in accordance with the provisions of Part II, 4.1.2. It is recognized that in certain areas of the world receivers designed for 50-kHz or 100-kHz channel spacing may continue to be used.

Also, airborne receivers operating in the 117.975-136 MHz will be required to reject adjacent channels (Reference 5).

4.7.2.3. Adjacent channel rejection. The receiving function shall ensure an effective adjacent channel rejection as follows:

a. 25-kHz channel spacing environment: 50 dB or more at plus or minus 25 kHz with respect to the assigned frequency and 40 dB or more at plus or minus 17 kHz;
b. 50-kHz channel spacing environment: 50 dB or more at plus or minus 50 kHz with respect to the assigned frequency and 40 dB or more at plus or minus 35 kHz;
c. 100-kHz channel spacing environment: 50 dB or more at plus or minus 100 kHz with respect to the assigned frequency;

4.7.2.4. RECOMMENDATION - Whenever practicable, the receiving system should ensure an effective adjacent channel rejection characteristic of 60 dB or more at

plus or minus 25 kHz, 50 kHz and 100 kHz from the assigned frequency for receiving systems intended to operate in channel spacing environments of 25 kHz, 50 kHz and 100 kHz respectively.

The Federal Communications Commission has provided for the conversion of the 117.975-136 MHz band to 25-kHz spacing. The FCC Rules and Regulations have been changed to reflect this change, as discussed by FCC in the following paragraph. 6

23. In summary, we are amending the rules to provide for a virtual doubling of the frequencies in the aeronautical mobile (R) band 117.975-136 MHz by providing for 25-kHz channel spacing. We are changing the frequency tolerances of both new ground transmitting and new airborne transmitting equipment to 0.002 and 0.003 percent respectively. The emission 13A9 will be authorized throughout the operational control band. We are establishing a cut-off date (1-1-74) for the type acceptance of new ground and airborne transmitters but not for the utilization of existing equipment. We are leaving this docket open pending the finalization of FAA's implementation plan and of proven evidence that continued use of 50-kHz configured equipment in the system causes unacceptable interference to the aeronautical community. We are providing the 25-kHz frequencies and the new tolerances to make it possible for manufacturers, station operators and users to proceed with the improvement of the system. We are providing for amendments which will allow better coordination and fuller usage of the flight test frequencies. In our opinion these amendments will allow for an assist in improvements to the existing aeronautical mobile (R) service in a manner most suitable to

⁶Federal Communication Commission Report and Order Regarding Aeronautical Mobile (R) VHF Band, Federal Register, October 19, 1973.

the early needs of air carriers to modify the offset carrier system, the financial needs of general aviation for adequate amortization time, and the FAA's implementation of the air space above 18,000 feet. Lacking a definitive FAA implementation program it would be unreasonable for the Commission to impose cut-off dates on aircraft operators which quite conceivably, would result in a considerable expenditure for new equipment of no improved value for lack of improved ground equipment with which to communicate.

The paragraphs quoted above indicate that conversion to equipment with 25-kHz selectivity is not mandatory at any specific future time in the US, and that conversion is required by ICAO only in certain unspecified areas of the world. However, information has been provided to ECAC indicating that Canadian commercial aircraft will be converted to the 25-kHz selectivity in accordance with the ICAO schedule, i.e., not later than January 1, 1977. Information provided to ECAC by ARINC indicates that, in the US, nearly all commercial aircraft are expected to be converted to 25-kHz selectivity not later than August 1977. Most ARINC ground stations in the 131.4-132 MHz band have been converted to 25-kHz selectivity at the present time; conversion of all stations in this band is expected to be complete by January 1977.

⁷TELCONS between J. Preis, ECAC, Ted Page, FAA, and Wayne Longman, Canadian Ministry of Transport, Ottawa, August 4, 1975.

⁸TELCONS between J. Preis, ECAC and R. Sollien, ARINC, July 8, 1975 and October 9, 1975.

The airborne transceivers are ARINC characteristic $546,^9$ $566,^{10}$ and $566A^{11}$ units. They transmit a 6A3 modulated 25-50 watt carrier at an exact multiple of 50 kHz, or 25 kHz \pm 6 kHz, or \pm 4 kHz stability respectively. They can receive 6A3 modulated signals above -133.5 dBW (133.5 dB below 1 watt) offset by as much as 8 kHz from an even multiple of 50 kHz (or 25 kHz).

The ground transceivers are similar to the airborne transceivers except:

- 1. Transmitter power varies from 5 watts to 1000 watts.
- 2. The transmitted carrier frequency may be offset by as much as 8 kHz from an exact multiple of 50 kHz (or 25 kHz). In the operational control band in the US, offsets up to 8 kHz are used. No offsets are used in the 125.4 to 126.0 MHz band in the US and Canada except in Alaska. In areas where offsets are used the stability is assumed to be 0 kHz.
- 3. In areas where offsets are not used the stability is \pm 6 kHz for 50-kHz channels \pm 3 kHz for 25-kHz channels.

⁹Airborne VHF Communications Transceiver System, ARINC Characteristic No. 546, Aeronautical Radio, Inc., October 1961.

Airborne VHF Communications Transceiver and Mark 1 VHF SATCOM Systems, ARINC Characteristic No. 566, Aeronautical Radio, Inc., October 17, 1968.

¹¹Mark 3 VHF Communications Transceiver, ARINC Characteristic No. 566A, Aeronautical Radio, Inc., August 23, 1972.

SECTION 3

ANALYSIS

POTENTIAL INTERFERENCE SITUATIONS

There are 7 potential interference situations. FIGURES 4 through 6 show the situations where the terrestrial system potentially interferes with the AEROSAT system. FIGURES 7 through 10 show the situations where the AEROSAT system potentially interferes with the terrestrial system. In all 7 figures, a desired signal path is designated with a D and an undesired signal path is designated with a U.

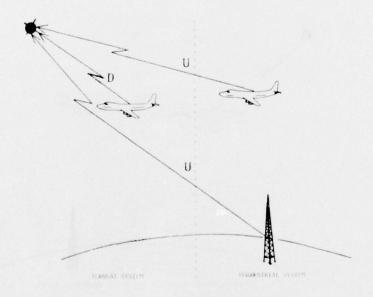


FIGURE 4. TERRESTRIAL SYSTEM POTENTIAL INTERFERENCE TO THE SATELLITE RECEIVER.

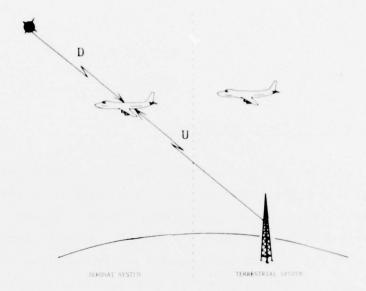


FIGURE 5. TERRESTRIAL SYSTEM GROUND TRANSMITTER POTENTIAL INTERFERENCE TO THE AEROSAT AVIONICS RECEIVER.

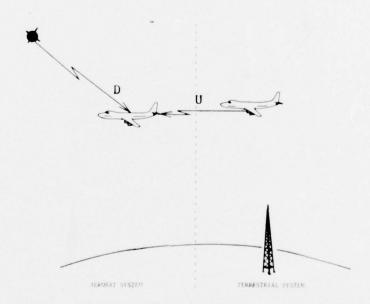


FIGURE 6. TERRESTRIAL SYSTEM AIRBORNE TRANSMITTER POTENTIAL INTERFERENCE TO THE AEROSAT AVIONICS RECEIVER.

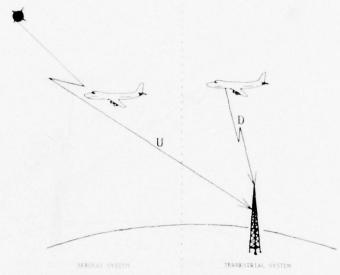


FIGURE 7. AEROSAT SATELLITE TRANSMITTER POTENTIAL INTERFERENCE TO THE TERRESTRIAL SYSTEM GROUND RECEIVER.

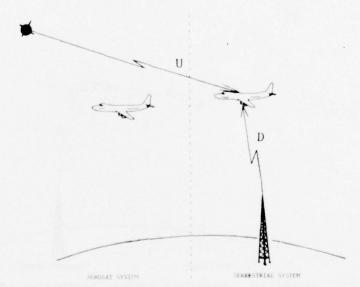


FIGURE 8. AEROSAT SATELLITE TRANSMITTER POTENTIAL INTERFERENCE TO THE TERRESTRIAL SYSTEM AIRBORNE RECEIVER.

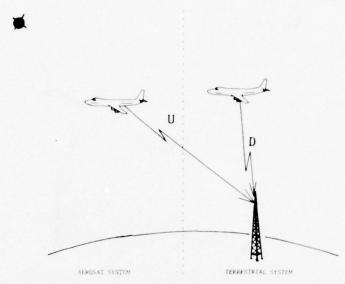


FIGURE 9. AEROSAT AIRBORNE TRANSMITTER POTENTIAL INTERFERENCE TO THE TERRESTRIAL SYSTEM GROUND RECEIVER.

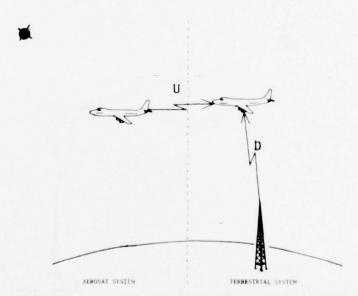


FIGURE 10. AEROSAT AIRBORNE TRANSMITTER POTENTIAL INTERFERENCE TO THE TERRESTRIAL SYSTEM AIRBORNE RECEIVER.

ANALYSIS PARAMETERS

General

This section provides the mathematical development of the factors (such as interference thresholds and propagation factors) used in the interference calculations.

Antennas and Transmission Lines

All antennas in the terrestrial system are vertically polarized and designed to be omni-directional (except for special cases which will be noted). The ground antennas have some gain which is assumed to be balanced by line losses. The terrestrial system airborne antennas have 0-dBi gain and the transmission line loss is assumed to be 1 dB.

All antennas in the AEROSAT system will be left-hand circularly polarized. The satellite antenna will have a gain of 10 dBi over the entire area of coverage. For the airborne antenna, AEROSAT will use the BOEING 747 SATCOM Antenna. This antenna is designed to give maximum gain for LH circularly polarized signals in all directions with an elevation of 10° or more (the gain is -2 dBi to +5 dBi over that region) and minimum gain elsewhere. The transmission line losses and transmitter powers are such that 19 dBW is delivered to the airborne antenna and 15 dBW is delivered to the satellite antenna.

For calculating interference between the AEROSAT system and the terrestrial system, the vertical polarization characteristics of the BOEING antenna are needed. FIGURES 11 and 12 give the upper and lower bounds of measured antenna voltage gain for all types of linear polarization. It is impossible to determine where between the upper and lower

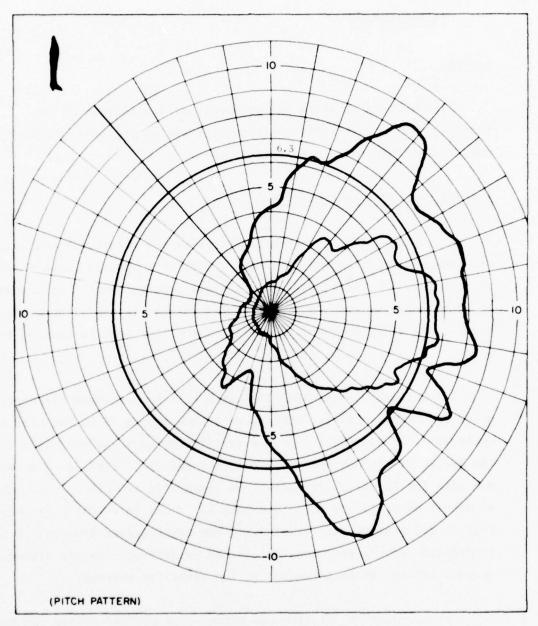


FIGURE 11. BOEING 747 SATCOM ANTENNA VOLTAGE PATTERN FOR LINEAR POLARIZATION (PITCH PATTERN).

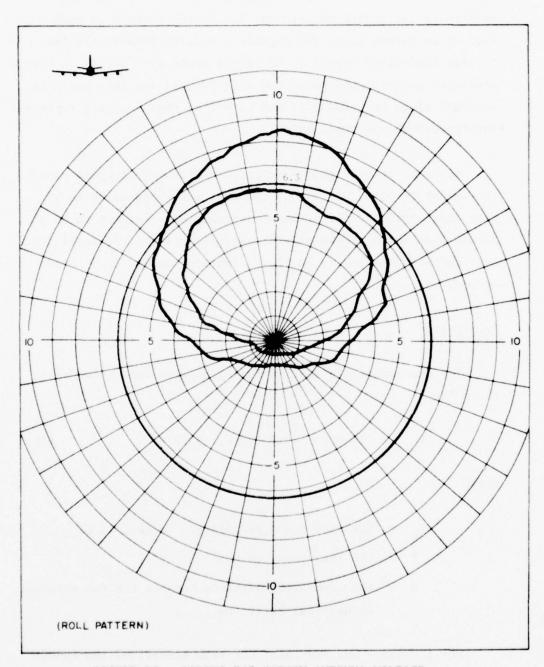


FIGURE 12. BOEING 747 SATCOM ANTENNA VOLTAGE PATTERN FOR LINEAR POLARIZATION (ROLL PATTERN).

bounds the vertical polarization pattern would be. Since the vertical polarization pattern for the Boeing antenna is only used to calculate gains for signals traveling between the two systems (undesired signals), the upper bound will result in higher predicted interference power and will be used for this analysis. The dark rings in FIGURES 11 and 12 with a radius of 6.3 represent isotropic 0-dB reference gain.

Since antenna heights between 10 and 1000 feet would yield equivalent results, all ground antennas were assumed to be 50 feet above ground. All airborne antennas were assumed to be at 40,000 feet above ground unless otherwise noted. All satellite antennas were assumed to be 19,400 nautical miles (nmi) above ground.

Propagation

The free-space propagation factor between two antennas is given by:

$$L_p = -20 \log f - 20 \log D - 37.8$$
 (1)

where

 $L_{\rm p}$ = the free-space propagation factor, in dB

f = the frequency, in MHz

D = the separation distance between the two antennas, in nautical miles (nmi).

For f = 125.7 MHz and 131.7 MHz (the centers of the AEROSAT VHF bands) the values of $L_{\rm p}$ become:

$$L_p$$
 at 125.7 MHz = -79.8 - 20 log D
 L_p at 131.7 MHz = -80.2 - 20 log D.

For communication with a geostationary satellite, the distance D varies from 19,400 nmi to 22,000 nmi depending on the satellite elevation angle. Therefore $\rm L_p$ becomes:

$$L_{\rm p}$$
 satellite-to-earth (125.7 MHz) = -165.6 dB to -166.6 dB $L_{\rm p}$ earth-to-satellite (131.7 MHz) = -166.0 dB to -167.0 dB.

Ionospheric Effects

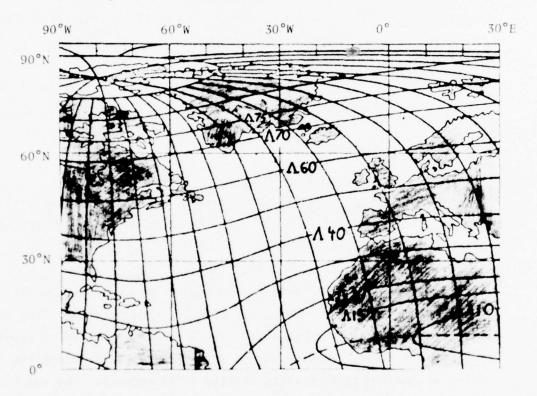
The ionosphere is a plasma of free electrons and nuclei with a magnetic field. The refractive index of a plasma is a function of the charge density. 12

This charge density is not a constant. The ionosphere contains many "clouds" of higher charge density that cause perturbations in the signals passing through them. These clouds, which often reach a length of several hundred yards, act like lenses, focusing the waves on certain regions while shading other regions. The exact effects these clouds have depends on their shape, size, number, and the magnetic field present. Due to the different refractive indices encountered, waves may be delayed at times causing unusual phase effects.

These perturbations are called scintillations. The frequencies of the scintillations range from one cycle per second to two minutes per cycle. The magnitude varies depending on location and time of

¹²Stratton, J. A., Electromagnetic Theory, McGraw-Hill, New York, 1941.

day. Scintillation is usually worst around midnight. Since scintillation is a function of magnetic field, it is not surprising that scintillation intensity is dependent on invariant geomagnetic latitude. Invariant geomagnetic coordinates are a function of the magnetic poles and are shown in FIGURE 13.



A denotes invariant geomagnetic latitude.

FIGURE 13. MAP WITH INVARIANT GEOMAGNETIC COORDINATES.

There are five scintillation regions. The VHF scintillation levels in the five regions are given in TABLE 1. The presently available measured data displays large variations. Therefore, only approximate values can be assigned at this time for consideration in this analysis.

TABLE 1

VHF SCINTILLATION LEVELS (APPROXIMATE)

Region	Latitudes	Signal Level Relative to the Median Exceeded 1% of the Time	Signal Level Relative to the Median Exceeded 99% of the Time
Polar	75° - 90°	5 dB	- 4 dB
Irregularity	60° - 75°	6 dB	- 5 dB
Boundary	50° - 60°	3 dB	- 2 dB
Mid-Latitudes	15° - 50°	<.5 dB	>5 dB
Equatoria1	0° - 15°	.5 dB	- 4 dB

FIGURE 14 shows continuous curves that approximate the data in TABLE 1.

A more complete discussion of scintillation may be found in the literature 13 , 14 from which the information used in the preceding discussion was obtained.

Multipath

In addition to the line-of-sight path between transmitter and receiver, the signal may simultaneously travel along a second path that reflects off of the earth's surface as shown in FIGURE 15.

This phenomenon is called multipath. Since the two paths do not have the same length, the signals may not be in phase. Depending on the phase, the reflected wave may reinforce or cancel the direct wave.

¹³Giordano, J. A., *Ionospheric Scintillations*, MTR-6559, Mitre Corporation, December 3, 1973.

¹⁴Elkins, T. J., and Slack, F. F., Observations of Traveling Ionospheric Disturbances Using Stationary Satellites, Journal of Atmospheric and Terrestrial Physics 1969, Vol. 31, pp. 431-439.

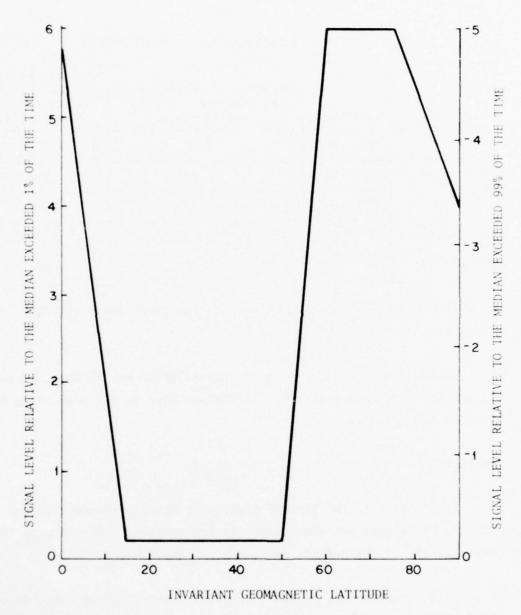


FIGURE 14. SCINTILLATION LEVELS.

To model multipath, what happens when the signal reflects off the earth's surface needs to be known. This is described by the reflection coefficient (R) of the earth's surface and by the divergence factor (Df) due to the curvature of the earth. The divergence factor is given by:

$$Df = \left[1 + \frac{2 r_1 r_2}{a (r_1 + r_2) \sin \theta}\right]^{-\frac{1}{2}}$$
 (2)

where

Df = the divergence factor

 r_1 = the distance from the transmitter to the point of reflection

 r_2 = the distance from the receiver to the point of reflection

 θ = the angle between the ray and the reflecting surface (grazing angle)

a = the radius of curvature of the surface.

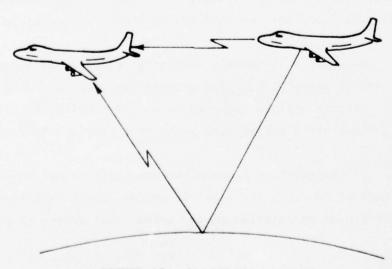


FIGURE 15. MULTIPATH DIAGRAM.

Section 3

Due to the high reflection coefficient of saltwater, maximum multipath effects occur when the reflecting surface is sea water. The reflection coefficient for sea water at 130 MHz is given in FIGURE 16. The upper curve ($\rm R_{\sc H}$) represents the magnitude of the reflection coefficient for horizontally polarized signals. The lower curve ($\rm R_{\sc V}$) represents the magnitude of the reflection coefficient for vertically polarized signals.

The magnitude of the reflected field at the receiver is given by:

where

|Er| = the magnitude of the reflected field

|Ei| = the magnitude the field would have had if it
had traveled the same path but reflected off
a flat, perfectly reflecting surface.

The field at the receiver is the algebraic sum of the direct path field and the reflected field. Maximums and minimums occur when the fields are in phase and 180° out of phase, respectively.

The propagation (including multipath) between two antennas below the ionosphere was modeled. The propagation loss including antenna and multipath factors will be referred to as the coupling factor in this report. There are 3 propagation paths that are important to this analysis:

1. Propagation between two omni-directional antennas, one at an altitude of 50 feet, the other at 40,000 feet. This represents desired air-ground propagation in the terrestrial system as shown in

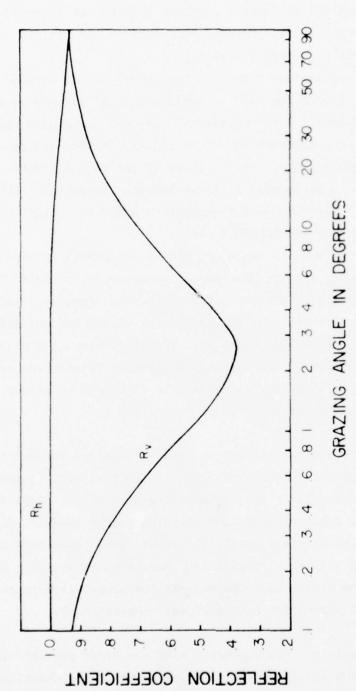


FIGURE 16. REFLECTION COEFFICIENT OF SEA WATER AT 130 MHz.

FIGURES 7, 8, 9, and 10. FIGURE 17 shows the coupling factor (CF) as a function of the horizontal distance between the antennas for this case. FIGURE 17 will be used to calculate all desired signal strengths in the terrestrial system.

- 2. Propagation between a Boeing 747 SATCOM antenna at an altitude of 40,000 feet and an omnidirectional antenna at an altitude of 50 feet. This represents undesired air-ground propagation between the terrestrial system and the AEROSAT system as shown in FIGURES 5 and 9. FIGURE 18 shows the coupling factor as a function of the horizontal distance between antennas for this case. FIGURE 18 will be used to calculate undesired signal strengths for the cases shown in FIGURES 5 and 9.
- 3. Propagation between a Boeing 747 SATCOM antenna and an omnidirectional antenna both at altitudes of 30,000 to 40,000 feet but not necessarily at the same altitude. This represents undesired air-air propagation between the terrestrial system and the AEROSAT system as shown in FIGURES 6 and 10. FIGURE 19 shows the coupling factor as a function of the horizontal distance between the antennas for this case. FIGURE 19 will be used to calculate undesired signal strengths for the cases shown in FIGURES 6 and 10.

These three figures include antenna effects and represent the coupling between the antenna terminals of the transmitting antenna and the antenna terminals of the receiving antenna. In each figure, the upper curve represents maximum signal strength (reflected signal in phase with the direct signal), the lower curve represents minimum signal strength (reflected signal 180° out of phase with the direct signal), and the middle curve represents the propagation factor for the direct path signal (reflected signal nonexistence).

This analysis will be concerned with the worst possible interference where the desired signal is minimum while the undesired signal

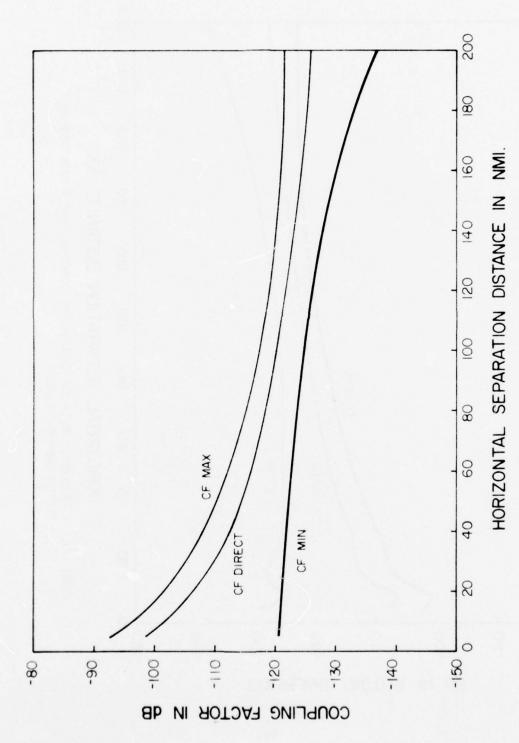


FIGURE 17. AIR-GROUND PROPAGATION IN THE TERRESTRIAL SYSTEM.

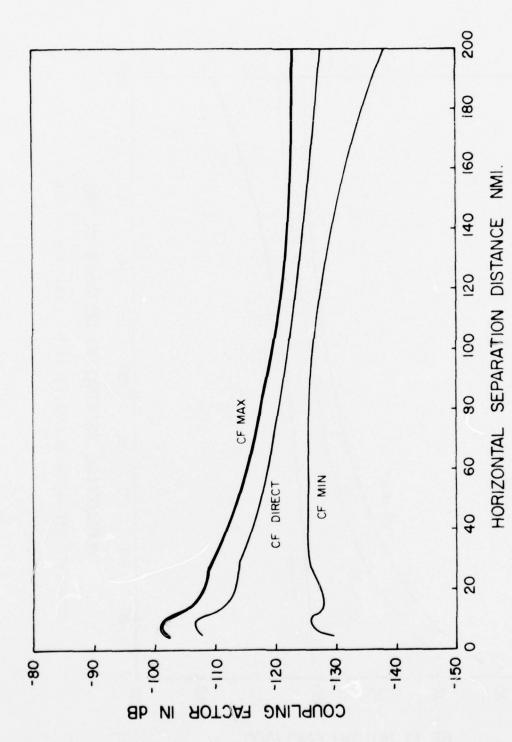


FIGURE 18. PROPAGATION BETWEEN AN AEROSAT AIRPLANE AND A TERRESTRIAL SYSTEM GROUND STATION.

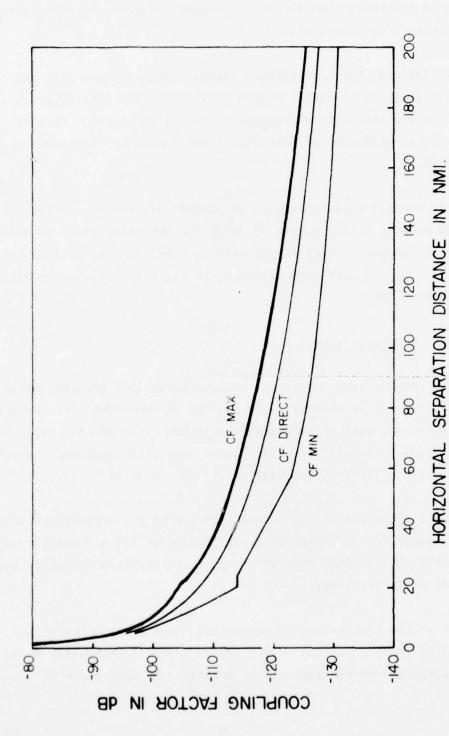


FIGURE 19. PROPAGATION BETWEEN AN AEROSAT AIRPLANE AND A TERRESTRIAL SYSTEM AIRPLANE.

is maximum. Therefore the lower curve in FIGURE 17 and the upper curves in FIGURES 18 and 19 are of primary interest. These curves are darkened for convenience.

In addition to the reflected signal being in phase with the direct signal, all possible airplane altitudes are considered in FIGURE 18 and the maximum propagation factor is picked. This is important at distances of less than 5 nmi (which are included in FIGURE 19).

For desired signals between an AEROSAT aircraft and a satellite, the MOU defines a link margin of 11 dB for "Scintillation, Multipath, and System Margin." This figure will be used, in lieu of calculating scintillation and multipath minimums, in all calculations for which it is applicable.

Receiver Characteristics

AEROSAT Receiver Characteristics. The AEROSAT receiver noise temperature will be at most 1,100 K. The IF bandwidth (for the airborne receiver) will be approximately 20 kHz. The AEROSAT voice and data demodulators will use phase locked loop (PLL) detectors, which will provide selectivity in addition to the IF selectivity.

While the specific PLL planned for use in the AEROSAT system was not available for analysis, an understanding of how a phase locked loop works will provide some of its performance characteristics in the presence of interference.

It is first necessary to understand how a Voltage Controlled Oscillator (VCO) generates FM. A VCO is an oscillator whose frequency is proportional to the controlling voltage. By applying a DC voltage to the VCO, a constant frequency "carrier" output is obtained. By superimposing an audio signal on the DC input voltage to the VCO, a frequency modulated carrier output is obtained. A VCO is essentially an FM modulator.

A typical PLL is diagrammed in FIGURE 20. The output of the VCO is compared to the PLL input signal be a phase detector. The output of the phase detector is "fed back" into the VCO through a filter. The PLL locks on to the input signal and the output of the VCO equals the PLL input signal. In effect the VCO is mimicking the transmitter that originally transmitted the signal. Therefore the input to the VCO must also mimic the input to the transmitter that originally transmitted the signal is the output of the PLL.

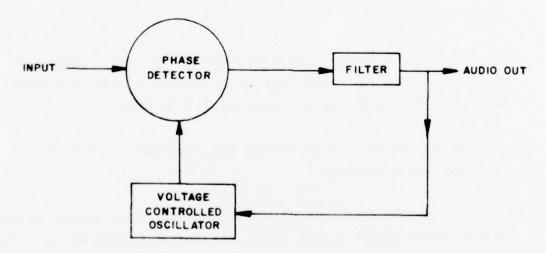


FIGURE 20. PHASE LOCKED LOOP DETECTOR.

The filter is an audio filter, but because of its location in the detector, it acts like an IF filter. The VCO and phase detector act like an oscillator-mixer with an output IF center frequency of zero Hz. For example if the filter is a low pass filter with a 3-dB point of 3 kHz, it would act like an IF filter that attenuates signals, 3-kHz off tune, 3 dB (i.e. a 3-dB IF bandwidth of 6 kHz). Very high selectivity is possible with a phase locked loop, but being a feedback system, the higher order filters necessary for high selectivity cause stability problems. The filter in the AEROSAT VHF receiver will have a transfer function like (Reference 3):

$$\frac{T + 1}{S} \tag{4}$$

where

s = complex frequency

T = a constant.

A PLL with a filter with the above transfer function is "second order."

The VCO in a PLL has a maximum range of frequencies over which it can vary. The PLL cannot lock on the signals outside this range, hence, the frequency range is named the lock-in range.

The PLL detector in the AEROSAT VHF receivers will have the following characteristics:

- 1. The PLL will be sensitive to interfering signals at the same frequency as the desired signal (co-channel interference). A co-channel undesired signal at the same level as the desired signal will cause consistent loss of lock and therefore makes communication impossible. 15
- 2. An interfering signal that is not co-channel but is inside the lock in range will be capable of pulling the PLL away from

¹⁵Britt, C. V. and Palmer, D. F., Effects of CW Interference on Narrow-Band Second-Order Phase Lock Loops, Research Triangle Institute, Research Triangle, NC, July 27, 1966.

the desired signal so that the PLL locks on to the undesired signal. It is not known how strong an undesired signal is necessary to make the PLL switch lock.

3. An interfering signal that is outside the lock in range will not break the lock of the PLL but may still cause some distortion.

Loss of lock is the most serious degradation a PLL can suffer. If a PLL is not locked on to a signal, it is not demodulating that signal.

The AEROSAT data demodulator will be a PLL with characteristics similar to the voice demodulator.

Terrestrial System Receiver Characteristics. An airborne receiver in the terrestrial system is the receiver part of an ARINC characteristic 546, 566, or 566A transceiver (References 9, 10, and 11). The sensitivity is such that a carrier with a power of -133.5 dBW, with a 1,000-Hz tone modulating it 30% applied to the antenna terminals will give a signal-to-noise ratio (S/N) at the speaker terminals of 4.74 dB. [(S+N)/N=6 dB]

The 546 and 566 characteristic transceivers each have two different IF selectivity characteristics (broad and narrow). The selectivity may be changed by interchanging plug-in filters, or, in some units, by throwing a switch. The broad IF selectivity for both transceivers is such that:

- 1. The 6-dB bandwidth is at least 30 kHz,
- 2. The 60-dB bandwidth is at most 63 kHz,
- 3. The 100-dB bandwidth is at most 80 kHz.

The narrow IF selectivity for the 546 characteristic transceiver is such that:

- 1. The 6-dB bandwidth is at least 13 kHz,
- 2. The 60-dB bandwidth is at most 37 kHz,
- 3. The 100-dB bandwidth is at most 47 kHz.

The narrow IF selectivity for the 566 characteristic transceiver is such that:

- 1. The 6-dB bandwidth is at least 16 kHz,
- 2. The 60-dB bandwidth is at most 30 kHz,
- 3. The 100-dB bandwidth is at most 37 kHz.

The 566A characteristic transceiver has only a narrow IF selectivity which is such that:

- 1. The 6-dB bandwidth is at least 16 kHz,
- 2. The 60-dB bandwidth is at most 34 kHz,
- 3. The 100-dB bandwidth is assumed at most 50 kHz.

The broad selectivity is used for 50-kHz channels. The narrow selectivity will be used for the 25-kHz channels.

The audio frequency response for all three receivers is such that:

- 1. From 300 Hz to 2,500 Hz, the response is essentially flat.
 - 2. There is sharp cutoff below 300 Hz,
- 3. There is a sharp cutoff above 2,500 Hz with frequencies above 5,750 Hz attenuated at least 20 dB.

Section 3

The characteristics of the ground receivers in the terrestrial system meet or exceed the characteristics of the airborne receivers.

All receivers in the terrestrial system are tuned to the center of the channel they are receiving even if the transmitter is offset from the center of the channel.

Terrestrial System Receiver Degradation Thresholds. This analysis will be concerned with the performance of the terrestrial system receivers in the presence of an interfering signal from an AEROSAT transmitter whose carrier frequency is 25 kHz away from the center of the channel being received. For all types of terrestrial system receivers, an undesired signal, 25 kHz away from the center of the channel being received and stronger than -63 dBW, (References 9, 10 and 11) may cause audible cross modulation. Since the narrow selectivity rejects signals below -63 dBW to below the noise level, the cross-modulation specification is the limiting factor for interference to terrestrial system receivers using the narrow selectivity.

Note that at -63 dBW, cross modulation is only "audible." It is not known at what undesired signal level, cross modulation becomes so severe that the desired communication is unintelligible.

An ECAC computer program, which models receiver performance in the presence of interference, was used to determine the degradation thresholds for the receivers using the broad selectivity. The computer program considers:

- 1. The type of receiver (AM, FM, PM, SSB, or FDM),
- 2. The IF bandwidth,
- 3. The IF selectivity characteristics,
- 4. Audio frequency response and rolloff characteristics,
- 5. The type of interference (AM, FM, PULSE, NONE, etc.),

- 6. The modulation on the interference (1000 Hz tone, 2000 Hz tone, voice shaped noise, no modulation, etc.),
 - 7. The RF signal to noise ratio,
 - 8. The RF signal to interference ratio,
- 9. The frequency separation between the signal and the interference.

Outputs of the program are:

- 1. The output signal-to-noise ratio at the speaker terminals,
- 2. The articulation index (AI).

Articulation index is a number between 0 and 1 that represents how well the desired communication can be understood. It differs from articulation score (AS) which is the average percent score that an average listener would get on a test where the listener hears a broadcast of random words (i.e. not sentences) and tries to identify the words. Note that for words in sentences, understanding would be higher.

AI is derived by purely mathematical means while AS is calculated from listener tests. An AS of 50% means that communication is such that only 50% of the words are understandable while an AI of 0.5 only means communication is better than for an AI of 0.4 and worse than for an AI of 0.6. However, for a particular type of interference, the relationship between AS and AI is approximately constant. The relationship between AS and AI for an FM voice signal interfering with AM is shown in FIGURE 21.16

DPSK interference on AM communications produces annoying sounds that cause listener fatigue, but the AS is much higher than for FM

¹⁶Kravitz, F., Communications/Electronics Receiver Performance Degradation Handbook, ESD-TR-73-014, ECAC, Annapolis, MD, July 1973.

voice interference on AM. Therefore AEROSAT voice communications would cause more interference to the terrestrial system than AEROSAT data communications would under the same conditions.

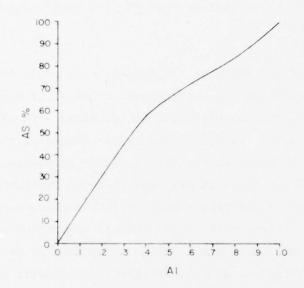


FIGURE 21. ARTICULATION SCORE AS A FUNCTION OF ARTICULATION INDEX, FOR FM VOICE OR AM VOICE.

Two types of receivers were modeled on the computer. An ARINC characteristic 546 and 566 transceiver using the broad selectivity was modeled, and since, in some areas of the world, some receivers don't meet the ARINC specifications, a "worst-case" receiver was also modeled.

The specifications for the worst-case receiver were the same as the ARINC characteristics except:

- 1. The sensitivity is -123 dBW (instead of -133.7 dBW),
- 2. The IF 6-dB bandwidth is 40 kHz,

- 3. The IF 60-dB bandwidth is 126 kHz,
- 4. The IF 100-dB bandwidth is 272 kHz.

The sensitivity characteristic is based on international regulations (Reference 5). The selectivity is that of three double-tuned IF stages with a bandwidth of 40 kHz. The ARINC characteristic broad selectivity is approximately that of five double-tuned IF stages with a bandwidth of 30 kHz.

A 4.74-dB S/N ratio at the speaker terminals, in the case where there is no interference, corresponds to an AI of 0.5. However, an experienced pilot can understand ATC messages 100% of the time at an AI of only 0.3.17 The 0.5 AI figure was chosen as the threshold for acceptable communication for this analysis to accommodate inexperienced listeners and non ATC messages used in the operational control system. The case of an interfering signal, with the characteristics of AEROSAT voice communications with 25-kHz channel separation was modeled. Curves of undesired signal strength (U) as a function of desired signal strength, (D) for a constant AI of 0.5, for the two receivers are given in FIGURE 22. These curves will be referred to as the degradation threshold curves for the rest of this report. Due to cross modulation, the maximum undesired signal strength that can be tolerated is -63 dBW. The degradation threshold curves include this -63 dBW maximum but for that part of the curves the AI is not necessarily 0.5. The carrier offsets used in the terrestrial system do not affect the degradation thresholds.

CALCULATIONS

The Necessity of Interleaving

With terrestrial system channels centered at multiples of 50 kHz, it has been proposed that the AEROSAT channels be centered at odd

¹⁷Gierhart, G. D., Hubbard, R. W., and Glen, D. V., *Electrospace Planning and Engineering for the Air Traffic Environment*, FAA-RD-70-71, US Department of Commerce, Boulder, CO, December 1970.

multiples of 25 kHz. This technique, which locates the AREOSAT channels between the terrestrial system channels, is called interleaving.

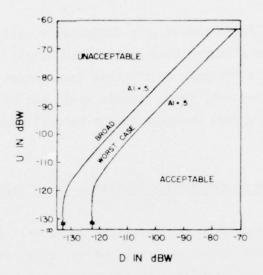


FIGURE 22. DEGRADATION THRESHOLD CURVES FOR THE CASE OF AEROSAT INTERFERENCE 25 kHz OFF TUNE TO TERRESTRIAL SYSTEM RECEIVERS.

The average signal strength of the satellite signal at the earth surface is given by:

Satellite EIRP = 25 dBW

Propagation Loss =
$$-166$$
 dB

 D_{avg} = -141 dBW

The average signal strength of a 25-watt signal (such as that radiated by an ARINC characteristic transmitter on an airplane) at a distance of 100 nautical miles, is given by:

Section 3

	Uavg	=	-107	dBW
Propagation Loss		=	-120	dB
Antenna Gain		=	0	dBi
Transmission Line	Loss	=	- 1	dB
Transmitter Output		=	14	dBW

The AEROSAT Phase Locked Loop Demodulator will not tolerate co-channel interference stronger than the desired signal. Therefore if the satellites were to transmit on the same frequencies as those used by the terrestrial system, their signals would not be received. Since the terrestrial system channels cannot be reallocated, interleaving is necessary.

Interference to the AEROSAT System by the Terrestrial System

Interference to the Satellite Receiver. FIGURE 4 shows the case where the terrestrial system is potentially interfering with the communication from the AEROSAT airborne transmitter to the satellite receiver. The ground stations and the airplanes as sources of interference are considered together because the distances are the same. This interference case is in the 131.4 to 132.0 MHz band.

The minimum desired-signal strength is given by:

Antenna Input Power	=	19	dBW
Minimum Airborne Antenna Ga	in =	- 2	dBi
Propagation Loss	=	-167	dB
System Margin	=	-11	dB
Satellite Antenna Gain	=	10	dBi
D _m	in =	-151	dBW

Section 3

System margin is the link margin defined in the MOU.

The maximum undesired signal strength (from transmitters in the US and Canada, to satellites over the Atlantic) is from a 500-watt transmitter at 131.8 MHz in Buffalo, New York. The maximum signal strength, $U_{\rm max}$ is given by:

	Umax	=	-124	dBW
Satellite Antenna	Gain	=	10	dBi
Polarization Loss		=	- 3	dB
Scintillation Maxi	mum	=	4	dB
Multipath Maximum		=	5	dB
Propagation Loss		=	-167	dB
EIRP		=	27	dBW

Therefore the minimum protection ratio, $(D/U)_{min}$, is given by:

$$(D/U)_{\min} = -27 \text{ dB}.$$

A typical transmitter would have a power of 50 watts giving:

$$U_{\text{max}} = -134 \text{ dBW}$$

and

$$(D/U)_{min} = -17 dB.$$

Since scintillation and multipath are rarely at a maximum simultaneously, the maximum undesired signal strength from a particular transmitter is rarely attained. The probability that the undesired signals from two transmitters would be at a maximum simultaneously is extremely low. Therefore, although there are many transmitters in each terrestrial

system channel, it is assumed that they do not combine to produce a higher undesired signal strength at a satellite than the highest undesired signal strength from any single transmitter.

The selectivity of the "satellite receiver" (see the discussion of the effective satellite receiver in the system description) must be sufficient to reject the maximum undesired signal strength from any transmitter in the 131.4 to 132.0 MHz band. By being offset 8 kHz from an exact multiple of 50 kHz, a ground transmitter carrier may be 17 kHz off tune from the center of an AEROSAT channel. However, all airborne transmitters and most US and Canadian ground transmitters with greater than 50-watt outputs are not offset. However, due to instability of up to ± 6 kHz, a transmitter that is not offset may be as little as 19 kHz off tune from the center of an AEROSAT channel. Outside the US and Canada, the incomplete data that has been supplied to ECAC indicates that offsets are not used in the 131.4 to 132.0 MHz band. This data also indicates that there are no undesired signals that exceed -124 dBW.

The selectivity requirements for the "satellite receiver" are different for different AEROSAT channels. APPENDIX A contains the US and Canadian data for use in calculating the selectivity requirements for specific AEROSAT channels. Assuming the satellite receiver will have the same selectivity characteristics for all channels, the selectivity requirements are:

- 1. Signals 19 kHz off tune at -124 dBW [(D/U $_{min}$) = -27 dB], and
- 2. Signals 17 kHz off tune at -128 dBW $[(D/U_{min})] = -23$ dB], must be rejected by the satellite receiver. In addition to the selectivity requirements, the satellite has a limited power budget. It may not be able to translate the undesired signals to the 5125 to 5250 MHz band and retransmit them without excessive power consumption.

Section 3

Interference to the Airborne AEROSAT Receiver. FIGURES 5 and 6 show the cases where the terrestrial system is potentially interfering with the communication from the satellite to the AEROSAT airborne receiver. These interference cases are in the 125.4 to 126.0 MHz band.

The minimum desired signal strength for both cases is given by:

Satellite EIRP
$$\approx$$
 +25 dBW

Propagation Loss \approx -166 dB

System Margin \approx -11 dB

Aircraft Antenna Gain \approx -2 dBi

 $=$ $=$ -154 dBW.

For the case where a ground transmitter is the potential source of interference, the maximum undesired signal strength is given by:

$$U_{\text{max}} = 10 \log P + CF_{\text{max}}$$
 (5)

where

P = the transmitter power, in watts

Therefore the protection ratio is given by:

$$(D/U)_{min} = -154 \text{ dBW} - 10 \log P - CF_{max}$$

where all terms have been defined previously.

Section 3

The most powerful terrestrial system ground transmitter (in the US and Canada) in the 125.4 to 126.0 MHz band is 100 watts (20 dBW) and the CF $_{\rm max}$ from FIGURE 18 is -102 dB so the minimum D/U is -72 dB (for the US and Canada).

For the case where a terrestrial system airborne transmitter is the source of interference, the analysis is identical except P=50 watts maximum minus 1-dB line loss.

Therefore:

$$(D/U)_{min} = -154 \text{ dBW} - (16 \text{ dBW}) - CF_{max}$$

= -170 dB - CF_max

where CF_{max} is the maximum coupling factor from FIGURE 19.

The minimum D/U that the AEROSAT VHF airborne receiver can tolerate is undefined. The minimum D/U that the AEROSAT VHF receiver will have to tolerate depends on how close to the 125.4 to 1260 MHz band terrestrial system the AEROSAT receiver will be operating. If an AEROSAT airplane were inside the area of coverage of a ground station operating in the band, then terrestrial system airplanes could get as close as 3 nmi to the AEROSAT airplane for an extended period of time. The CF_{max} between an AEROSAT airplane and a terrestrial system airplane at 3 nmi is -86 dB, resulting in a D/U of -84 dB. A D/U of -84 dB represents more interference than will be encountered due to any ground transmitter. Therefore -84 dB is the protection ratio required for AEROSAT to be able always to operate inside the area of coverage of a ground station. Since no offsets are used by airborne transmitters, the interfering signal will be at least 19 kHz off tune.

A receiver capable of rejecting an undesired signal only 19 kHz off tune and 84 dB above the desired signal level is very difficult

to build. Even if the selectivity is great enough to reject the signal, the front end may not be able to handle such a strong interfering signal. If the front end is designed so that the interfering signal does not overload, desensitize, produce intermodulation in, or otherwise disturb it, then the sensitivity may be compromised.

However, the D/U of -84 dB occurs rarely, and only when an AEROSAT airplane is inside the area of coverage of a terrestrial system ground station in the 125.4 to 126.0 MHz band. These coverage areas do not extend more than a few miles off shore (though they cover almost all land areas).

It is required that AEROSAT operate in all oceanic areas not covered by Extended Range VHF stations.

How much interference to the airborne AEROSAT receiver exists at the extreme limits of the Extended Range VHF coverage areas needs to be known. Since these stations are not in the 125.4 to 126.0 MHz band, the areas in which AEROSAT must operate do not overlap the areas where AEROSAT would experience severe interference. Maps showing the coverage areas of the Extended Range VHF stations in the US, Canada, and the North Atlantic Region are included in APPENDIX C. The maps in APPENDIX C show Extended Range coverage areas extending more than 200 nmi out over the ocean. The most powerful transmitter in the 125.4 to 126.0 MHz band (in the US and Canada) is 100 watts (20 dBW). The CF between a terrestrial system ground transmitter and an airborne AEROSAT receiver 200 nmi away is -123 dB, so the $U_{\rm max}$ is -103 dBW and the $(D/U)_{min}$ is -51 dB. The CF_{max} between a terrestrial system airborne transmitter and an airborne AEROSAT receiver 200 nmi away is -125 dB so the U_{max} from an airborne terrestrial system transmitter is -109 dBW and the $(D/U)_{min}$ is -45 dB. Since offsets up to 8 kHz are used in this band in Europe, the interfering signal may be only 17 kHz off tune.

For AEROSAT to be able always to operate in all oceanic areas not covered by Extended Range VHF stations, the AEROSAT VHF avionics receiver must be able to operate in the presence of undesired signals, 17 kHz off tune, 51 dB above the desired signal level. As mentioned earlier, for AEROSAT to be able always to operate over land, the AEROSAT VHF avionics receiver must be able to operate in the presence of interference 84 dB above the desired signal level.

These two requirements are the minimum and maximum requirements respectively. If the interference rejection capabilities of the AEROSAT VHF avionics receivers fall somewhere between the two requirements given above, then AEROSAT will always be able to operate less than 200 nmi away from the shore, but not always over land.

The receivers should be designed to reject as much interfering signal (17 to 19 kHz off tune) with a goal of being able to reject signals 19 kHz off tune, 84 dB above the desired signal level, without compromising sensitivity. To aid in achieving this design goal, the lock in range of the PLL should be within \pm 17 kHz of the center of the channel.

Interference to the Terrestrial System by the AEROSAT System

The Satellite as a Source of Interference. FIGURES 6 and 7 show the cases where the satellite is potentially interfering with the terrestrial system. This interference is in the 125.4 to 126.0 MHz band. The maximum strength of the satellite signal to terrestrial system receivers (when the two transmitters are combined for maximum power) is given by:

Section 3

Umax	=	-129	dBW.
Antenna Gain	=	0	dBi
Polarization Loss	=	-3	dB
Scintillation Maximum	=	6	dB
Multipath Maximum	=	6	dB
Propagation Loss	=	-166	dB
Satellite EIRP	=	28	dBW

The degradation threshold curves (FIGURE 22) show that for an undesired signal strength of -129 dBW, the desired signal strength that gives an AI of 0.5 is the same as for an undesired signal strength of $-\infty$ dBW. Therefore, the satellite signal is not a significant source of interference.

The Airborne AEROSAT Transmitter as a Source of Interference. FIGURES 9 and 10 show the cases where the airborne AEROSAT transmitter is potentially interfering with the terrestrial system. This interference is in the 131.4 to 132.0 MHz band.

For the case where the airborne AEROSAT transmitter is potentially interfering with a particular terrestrial system ground receiver that is listening to nearby airplanes, the minimum desired signal strength is given by:

$$D_{\min} = 13 \text{ dBW} + CF_{\min} \tag{7}$$

where

13 dBW = the minimum EIRP for an airborne terrestrial system transmitter (25 W power out minus 1 dB line loss)

Section 3

CF the minimum coupling factor from FIGURE 17 (lower curve) for any airplane in the area of coverage of the ground station.

 ${\rm CF}_{\min}$ is found by finding the distance to the farthest point away from the ground station in the area of coverage, and using FIGURE 17 to obtain ${\rm CF}_{\min}$.

Given the minimum desired signal strength (D_{\min}) , the maximum allowable undesired signal strength (U_{\max}) may be found by using:

- 1. The degradation threshold curves (FIGURE 22) for broad selectivity or worst-case selectivity receivers, or
 - 2. U = -63 dBW for the narrow selectivity receivers.

Since the ground receivers meet the ARINC characteristics, the lower (worst case) curve in FIGURE 22 is not used. Also:

$$U_{\text{max}} = CF_{\text{max}} + 19 \text{ dBW}$$
 (8)

where

19 dBW = the power delivered to the antenna (after line losses) by the airborne AEROSAT transmitter

 CF_{max} comes from the upper curve of FIGURE 18.

For a given value of U, $\mathrm{CF}_{\mathrm{max}}$ may be determined. Knowing $\mathrm{CF}_{\mathrm{max}}$, from FIGURE 18, a "minimum distance" can be obtained that is how far away from the ground station the AEROSAT airplane must be when it is transmitting. When the AEROSAT airplane is within the "minimum distance" of a particular ground station, its AEROSAT VHF transmitter must be off. The area where the VHF transmitter must be off will be referred to as the "interference area" for the rest of this report.

Section 3

For narrow selectivity ground receivers, $U_{max} = -63~dBW$, therefore $CF_{max} = -82~dB$. However FIGURE 18 shows a CF_{max} of -102 dB or less. Therefore, there are no interference areas due to narrow selectivity ground receivers.

Broad selectivity ground receivers can experience interference from airborne AEROSAT VHF transmitters. Around each broad selectivity ground receiver will be an interference area described by a circle whose radius is equal to the "minimum distance." These circles may be drawn for the locations of all ground receivers, whose channels are adjacent to a particular AEROSAT channel. A plot of these circles on a map will depict the areas where operation of airborne AEROSAT VHF transmitters in that particular channel may cause interference. However, this map only shows the areas that are interference areas due to interference to ground receivers trying to listen to nearby airplanes. Interference to airborne receivers trying to listen to ground stations will generate (by a similar process) even larger interference areas (in most cases).

In an interference area, the AEROSAT VHF transceiver on an airplane *must not* be used. In the cases where the terrestrial system may interfere with the AEROSAT system (discussed earlier), the AEROSAT VHF subsystem may be used but will occasionally experience interference.

For the case where the airborne AEROSAT transmitter is interfering with an airborne receiver that is listening to a nearby ground station, the minimum desired signal is given by:

$$D_{\min} = 10 \log P + CF_{\min}$$
 (9)

where

P = the power of the ground transmitter which the airborne receiver is listening to

Section 3

CF = the minimum propagation factor (from FIGURE 18) between the airplane and the ground station.

This is calculated for every possible airplane location in the area of coverage of every ground station.

Once again U is found from:

- 1. The degradation threshold curves for broad or worst case selectivity receivers, or
 - 2. U = -63 dBW for narrow selectivity receivers.

As before:

$$CF_{max} = U_{max} - 19 \text{ dBW}, \qquad (10)$$

but FIGURE 19 is used to determine the minimum distance.

For the narrow selectivity receivers, CF_{max} = -82 dB and the minimum distances is 1.8 nmi.

Although airplanes do occasionally approach each other to within 1.8 nmi, this distance is small enough that the AEROSAT interference to airborne terrestrial system receivers using the narrow selectivity is insignificant. There are 3 reasons for this conclusion.

- 1. At a separation distance of 1.8 nmi, cross modulation is only audible, not necessarily intolerable.
- 2. Airplanes flying in the same direction, and under ATC rules, are separated by at least 3 nmi. Airplanes flying in different directions would be in close proximity for only a few seconds.

3. Using all of the Collision Avoidance Systems, ¹⁸ in all possible instances, a separation distance of 1.8 nmi would be avoided

For the broad or worst-case selectivity receivers, the minimum distance obtained from FIGURE 19 will usually be significant. Once again charts showing interference areas can be made. However, the production of the charts for this case differs from the production of the charts for the previous case.

For this case, each possible airplane location must be considered and a circle of "minimum distance" radius drawn around that location. For each ground station, there will be a collection of circles. This collection of circles describes an "interference area" associated with the ground station. The dozens of transmitters in the channels adjacent to a particular AEROSAT channel yield a total interference area.

Finally, the interference areas from both preceding cases are combined to form the final interference area for each particular AEROSAT channel. To prevent interference to the terrestrial system, AEROSAT VHF transmitters should be prohibited from emitting in the interference area designated for each specific AEROSAT channel.

The Sufficiency of Interleaving

As mentioned earlier, it is required that AEROSAT operate in all oceanic areas not covered by Extended Range VHF stations. It is necessary to know if the interference areas will make the above requirement impossible in some areas.

Airborne Collision Avoidance System, Statement of Airline Policy and Requirements and a Description of the System, Air Navigation/Traffic Control Division, Air Transport Association of America ANTC Report No. NR. 117 Revision 10, 2 May 1971, corrected 27 September 1971.

In the gradual worldwide conversion to 25-kHz channel spacing, the US and Canada will be among the first to convert. Airplanes that may be flying in the US or Canada will be among the first airplanes to switch to the narrow selectivity. It is assumed that airplanes that fly as far out over the ocean as the outer regions of Extended Range VHF stations will also be among the first to switch to the narrow selectivity. Therefore it is not necessary for AEROSAT airplanes to avoid any of the Extended Range VHF coverage areas specifically.

The weakest terrestrial system ground transmitters encountered have an output of 5 watts. The coverage areas of such low power transmitters extend less than 50 nmi away from the ground stations. Therefore the CF_{min} (FIGURE 17) is -122 dB and the D is -115 dBW. From the degradation threshold curves (FIGURE 21), the maximum allowable U for the worst-case receivers is -109 dBW. Therefore, from Equation 10, the CF_{max} is -128 dBW, and from FIGURE 19 the minimum separation distance is approximately 230 nmi. Assuming that the coverage areas of the 5-watt terrestrial system transmitters are strictly over land, all areas more than 230 nmi offshore are not interference areas.

APPENDIX C includes the coverage areas of the Extended Range VHF transmitters in the US, Canada, and the North Atlantic region. These coverage areas generally extend at least 250 nmi out over the ocean.

Therefore, even in the worst case, the coverage areas of the Extended Range stations extend beyond the interference areas. This means 25-channel separation is sufficient and that the proposed interleaved candidate frequency plan will yield a workable system.

Interference in the US and Canada

In this subsection, the general results developed in the calculations subsection will be applied to the specific interference cases in the US and Canada. Each of the interference cases depicted in FIGURES 3 through 9 will be discussed. Planned satellite locations over the Pacific Ocean at 155° and 180° West longitude will be analyzed.

Interference to the AEROSAT Satellite

Receiver. This case, which is in the 131.4 to 132.0 MHz band, is shown in FIGURE 4.

As discussed in the calculations subsection, the maximum undesired signal strength at the satellites over the Atlantic Ocean from transmitters in the JS and Canada is -124 dBW. For use in specific calculations, a list of all ground transmitters in the US and Canada, in the 131.4 to 132.0 MHz band, which use omnidirectional antennas, is contained in APPENDIX A. This list includes maximum received power at satellites at 15°, 40°, 155°, and 180° West longitude.

There are 3 ground transmitters in the US and Canada in the 131.4 to 132.0 MHz band that use directional antennas.

1. Mount Haleakala, Hawaii, 20°42'42" North latitude, 156°15'33" West longitude. A 1,000-watt transmitter is coupled to a 17.2-dBi gain antenna with 0.5-dB line loss. The antenna beam has an elevation angle of 0°, an azimuth of 054°T, and a horizontal 3-dB beamwidth of 32°. The carrier frequency is 131.95 MHz. The azimuth to the satellite at 180° West longitude would be 231.2°T and the azimuth to the satellite at 155° West longitude would be 176.4°T, so the satellites would not be in the mainbeam of the antenna. The maximum sidelobe gain is assumed to be 0 dBi. Therefore, the maximum received signal strength at the Pacific satellites would be:

		-124	dBW.
Satellite Antenna Gain	=	10	dB
Polarization Loss	=	- 3	dB
Scintillation Maximum	=	0	dB
Multipath Maximum	=	5.7	dB
Propagation Factor	=	-166.1	dB
Antenna Gain	=	0	dB
Line Loss	=	5	dB
Transmitter Power	=	30	dBW

The Atlantic satellites would be over the horizon and therefore their received signal strengths can be neglected.

- 2. San Francisco, California, 37°30'23" North latitude, 122°22'16" West longitude. A 1,000-watt transmitter is coupled to a 21-dB gain antenna with a 2-dB line loss. The antenna beam has an elevation angle of 0° an azimuth of 252°T, a 3-dB horizontal beamwidth of 20°, and a vertical 3-dB beamwidth of 5° to 10°. The carrier frequency is 131.95 MHz. The azimuth to the satellite at 180° West longitude would be 248.9°T and the elevation angle would be 16.8°. The azimuth to the satellite at 155° West longitude would be 226.4°T and the elevation angle would be 34.8°. Due to the narrow vertical beamwidth, both satellites would not be in the mainbeam of the antenna and the maximum received signal strength at the Pacific satellites would be approximately the same as from Mount Haleakala. The Atlantic satellites would be over the horizon and their received signal strengths can be neglected.
- 3. Santa Ynez, California, 34°31'33" North latitude, 119°58'43" West longitude. A 1,000-watt transmitter is coupled to a 16-dBi gain antenna with 0.4-dB line loss. The antenna beam has an elevation angle of 0°, an azimuth of 245°T and a horizontal 3-dB beamwidth of 32°. The vertical beamwidth is the same as the horizontal except that since vertical polarization is used, an extra factor of:

$$\frac{\cos^2 (\pi/2 \sin E)}{\cos^2 E} \tag{11}$$

FAA-RD-77-56, I

Section 3

where

E = the elevation angle.

This factor is introduced due to the radiation pattern of the vertically polarized elements. The carrier frequency is 131.944 MHz. The azimuth to the satellite at 155° West longitude would be 231.0°T and the elevation angle would be 35.4°. Therefore the satellite would not be in the mainbeam of the antenna and the maximum received signal strength at the satellite would be approximately the same as from Mount Haleakala. The Atlantic satellites would be over the horizon and their received signal strengths can be neglected. The azimuth to the satellite at 180° West longitude would be 251.9°T and the elevation angle would be 16.0°. Consequently:

$$\frac{\cos^2 \left[\pi/2 \sin \left(16^{\circ}\right)\right]}{\cos^2 \left(16^{\circ}\right)} = .8917 = -0.5 \text{ dB}.$$

Therefore the antenna gain in the direction of the satellite would be 12.5 dB. The maximum received signal strength at the satellite at 180° West longitude would be:

Transmitter Power	=	30.0	dBW
Line Loss	=	-0.4	dB
Antenna Gain	=	12.5	dB
Propagation Factor	=	-166.9	dB
Multipath Maximum	=	5.2	dB
Scintillation Maximum	=	0.0	dB
Polarization Loss	=	-3.0	dB
Satellite Antenna Gain	11	10.0	dB
		and the second second second second	

-112.6 dBW.

This signal is 19 kHz off tune from the 131.925-MHz AEROSAT channel and is 31 kHz off tune from the 131.975-MHz AEROSAT channel. If an AEROSAT satellite cannot handle this interference without modification, then the satellite at 180° West longitude must be modified. There are three possible solutions.

- 1. Modify the satellite so that it will be able to handle the -112.6-dBW undesired signal (a band reject filter tuned to 131.944 MHz is a possible method).
- 2. Move the satellite at 180° West longitude out of the undesired signal beam. However to reduce the maximum undesired signal level to -124 dBW, the satellite must be located east of 170° West longitude or west of 155° East longitude. At 170° West longitude the satellite would be too close to the satellite at 155° West longitude and simultaneous eclipse (where both satellites are in the earth's shadow) would be possible.
- 3. Do not use the satellite at 180° West longitude for the channels at 131.925 and 131.975 MHz, (that is operate it in the 131.4 to 131.9 MHz band) and design the satellite so that it rejects signals above 131.9 MHz.

Interference to the AEROSAT Airborne Receiver. These cases, which are in the 125.4 to 126.0 MHz band, are shown in FIGURES 5 and 6.

In the interference to the AEROSAT system by the terrestrial system subsection, it was shown that depending on how much of the 17 to 19 kHz off-tune interference the AEROSAT VHF avionics receiver can tolerate, the AEROSAT VHF subsystem will be able to operate closer than 200 nmi away from the shore, or, over land. Since offsets are not used in the US and Canada in this band, and since 25-kHz channels will be implemented in the US and Canada by the time the AEROSAT system is operational, all interfering signals will be at least 21-kHz off tune

(due to \pm 4 kHz stability). Therefore, there will be less interference to the AEROSAT system near the US and Canada than near countries that use offsets and/or 50-kHz spacing. For general information purposes, a list of all ground transmitters in the US and Canada is included in APPENDIX B.

The Satellite as a Source of Interference. These cases, which are in the 125.4 to 126.0 MHz band, are shown in FIGURES 7 and 8. As shown in the calculations subsection, the satellites are not significant sources of interference.

These cases, which are in the 131.4 to 132.0 MHz band, are shown in FIGURES 9 and 10. Since 25-kHz channels will be implemented in the US and Canada by the time the AEROSAT system is operational there will be no interference areas (where the AEROSAT VHF airborne transceiver must not be used).

SECTION 4

RESULTS

RESULTS

- 1. For the AEROSAT VHF subsystem to meet the minimum requirement that it be able to operate in all oceanic areas not covered by Extended Range VHF stations without interfering with or experiencing interference from systems presently operating in the same band, the following conditions must be satisfied.
- a. The AEROSAT avionics receivers must be able to operate in the presence of undesired signals 17 kHz off tune and 51 dB above the minimum desired signal level.
- b. For the Atlantic system, the "effective satellite receiver" must be able to operate in the presence of undesired signals 27 dB above the minimum desired signal level and 19 kHz off tune, and in the presence of undesired signals 23 dB above the minimum desired-signal level and 17 kHz off tune. Both selectivity and the satellite power budget should be considered in meeting this requirement.
- c. For the Pacific system, the "effective satellite receiver" must meet all the Atlantic system requirements, and the "effective satellite receiver" at 180° West longitude must be able to operate in the presence of an undesired signal 38.4 dB above the minimum desired signal level, at 131.944 MHz.
- 2. For the AEROSAT VHF subsystem to exceed the minimum requirement discussed in Result 1, that is, to operate in the coverage area of an Extended Range VHF station, or overland, the following conditions must be satisfied.
- a. In areas where the terrestrial system receivers are not using the narrow selectivity, there will be "interference areas"

as described in the ANALYSIS section (under Interference to the Terrestrial System by the AEROSAT system). When an AEROSAT airplane is in an interference area, its AEROSAT VHF transceiver must not be used.

- b. In areas where the terrestrial system receivers are using the narrow selectivity, AEROSAT VHF communications may be used without any restrictions. However, depending on the ability of the AEROSAT VHF avionics receiver to operate in the presence of adjacent channel interference, the AEROSAT VHF satellite-to-aircraft communications may occasionally experience interference.
- c. For AEROSAT VHF communications never to experience interference for all possible locations of the AEROSAT airplane, the AEROSAT VHF avionics receiver must be able to operate in the presence of undesired signals, 19-kHz off tune, 84 dB above the minimum desired signal level.
- 3. In the US and Canada, the terrestrial system receivers will be using the narrow selectivity. Therefore, AEROSAT VHF communications may be used without any restrictions. However, unless the AEROSAT VHF avionics receiver is able to operate in the presence of undesired signals, 19-kHz off tune, 84 dB above the minimum desired signal level, AEROSAT VHF satellite-to-aircraft communications will suffer occasional interference.

APPENDIX A

TRANSMITTERS IN THE US AND CANADA IN THE 131.4 to 132.0 MHz BAND

TABLE A-1 is a list of all the ground based transmitters in the US and Canada in the 131.4 to 132.0 MHz band which use omnidirectional antennas. Received powers at satellite locations above 15° , 40° , 155° , and 180° West longitude are included in the list.

The column headings in TABLE A-1 are described below:

FREQ IN MHz - gives the exact carrier frequency of each transmitter including offset. The transmitters are listed by frequency.

CALL LETTERS - the first four characters of the call sign of each transmitter.

CITY - ground-station location.

STATE OR PROVINCE - ground-station location.

LATITUDE - ground-station location.

LONGITUDE - ground-station location.

TRANSMITTER POWER IN WATTS - ground-station transmitter power.

SATELLITE AT 15 DEG. W. - for the geostationary satellite at 15° West longitude, calculated values are given for the elevation angle from the ground station to the satellite (ELEV. ANGLE), received signal strength (RECEIVED POWER IN dBW) from both the ground station (GROUND) and an aircraft in the vicinity of the ground station (ACFT). The received signal strength is the maximum received signal strength at the satellite antenna terminals including:

a. The transmitter power - For the airborne transmitters, the maximum power of 50-watts is used.

- b. The transmitting antenna gain and line losses.
- c. The free-space propagation factor.
- d. Multipath for maximum signal strength, the factor is 20 $\log (1 + RD)$ where R is the reflection coefficient for sea water and D is the divergence factor.
- e. Scintille ion the values which are exceeded 1% of the time in TABLE 1 are used.
 - f. Satellite antenna gain.

The scintillation and multipaths are such that the "received power in dBW" is exceeded approximately 0.1% of the time. For elevation angles less than 10° , the elevation angles are not accurate due to atmospheric refraction and the received signal strengths are unpredictable and are therefore not shown.

SATELLITE AT 40 DEG. W. - similar to the satellite at 15° W., and similar values for the satellite at 40° West longitude. SATELLITE AT 155 DEG. W. - similar values for the satellite at 155° West longitude.

SATELLITE AT 180 DEG. W. - similar values for the satellite at 180° West longitude.

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TRANSMITTERS IN THE US AND CANADA IN THE 131.4 TO 132.0 MHz BAND (Page 1 of 9)

TABLE A-1

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| FLEV | | -15.47 | -11.24 | 16.36 | -18.32
 | -13.11 | -21.98 | 10.40 | 15.65 | 114.11
 | -20.36 | -15.38 | -15.18 | 70,01 | -19,52
 | 16.81 | -30.21 | 10,01 | -31.12 | 62,71
 | -12,92 | 19.67 | 10.10 | -13.35 | 54,67
 | -15.54 | -5.01 | 111.87 | 14.84 | -10.10
 | 000 | -20.47 | -20.10 | 14.17 | -15.30
 | -18.28 | -16.72 | -10.12 | -16.62 | -8.47 | -17.05
 | 00.00 | -8.83 | -9.26 | 16.77 | 110.11
 | -10.38 | -3.86 | -8.71 | -21.38 | |
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 | 7.45 | -4.24 | 8.07 | 10.00 | 04.0
 | 23.00 | 1.85 | 6.34 | 65.29 | 93
 | 34.79 | 7.67 | 0 0 | -8.63 | 41.21
 | 2.97 | 14.68 | 30.01 | 19.9 | 64.80
 | 2.67 | 14.14 | 30.90 | 34.62 | 2.05
 | 13.22 | -2.12 | -1.75 | 07.70 | 3.04
 | .32 | 1.26 | 17.0 | 2.05 | 40.6 | 5.07
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| POWER I | SROUND | -135. | | 1251 | -103.
 | -184. | -141. | -138. | |
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 | -134. | -147 | | -134. |
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 | 1.04. | -135. | -138. | -198. | -138.
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 | 28.90 | 31.95 | 22.36 | 20.10 | 20.00
 | 36.13 | 26.87 | 33.04 | -45.26 | 31.43
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 | 25.55 | 23,52 | 28.03 | 24.70 | -33.13
 | 32.78 | 18.33 | 3.30 | *6. | 21,33
 | 10.01 | 31.82 | 31.48 | 28.94 | 27.08
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TABLE A-1 (Page 2 of 9)

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04 74 7		POWER IN DB	GROUND	-133.	-142.	-135.	-142.							,	-137.	-137.	-137.	-137.	-138.	-137.				-142.	-141.	-134.			-138.	-137.	-128.	-137.		-143.	-140.	-135		-142.		-144.		-146			-142	-134.	-138.	-1.5	-130	100	100
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	FLEV	N.	-1.35	-19.	-18,32	14.			-3.6	-2.	•	-12,78	. 22.	4	1,10	-10	-13.		-15.6	-13.	-111.		-7.6	-11.	-1.	136.00	-12.4	-10,2	-15.	-12.7	-12.38	-11-		-14,78	04.6-			-15.	-13.34			.6.	-17.0	-10.0	9-1-	- 10	-13.78
. 930 C	POWER IN DRY	ACFT	-139.	. 66 .		-138.	-136.	-1 18.	-139.	-139.	-139.		100	-139.	-134.			-130					-139.						0.0	* * 6 * 1			-130.		-139.		-130			-129.	-1.39	-139.		-139.	-136		
SATELLITE AT 155 DEG	POWER	BROUND	-136.				-138.	-117.	-145.	-140.	-138.		-147.	-138.	-138.			-138					-145.									:	14.5		-145.		-138.			-149	-134.	-145.		-138.	-138.		
SATELLI	FLEV	IN DEG	15.20	86	192	34.09	20.57	24.40	13.28	13.83	15.62	6.63	26.93	11.58	17.62	3.68	000	26.00	2.81	7.65	8.63	***	13.85	9.41	5.26	60.	7.81	8.16	00.0	01.0	8.78	8.78	15.50	3.40	12.08	0.0	12.50	5.67	99.90	20.77	11.70	11.37	5.07	11.31	10.11	-2.71	7.40
* 9 3	VED	ACET	-134.	. 137	-137,		-117		.134.	-134.		-133.	-1113.	-133.		-136.	130	133	-136.	138.	138.	. 26.	-138:	138.	138.		138.	-136.	138.	138.	138.	138,		.138.	.138.	. 138	138.	.138.	-138.	138.	. 1 14	138.	138.	138.		133.	-138.
	POWER IN DAN	SROUND	-133.									-132.		-132.												-143.													-137.								-144.
SATELLITE AT 40 DES	NELE	9	11.04			. 2.																																									
		ACET IN		-138.	-138.		-1.10												-136.							-138.			-139,									-139.					-139.			-134.	
	POWER IN DBM	GROUND			-137.														-137.										-145.									-136.					-148.			-135.	
SATELLITE AT 15 DES	ELEV		-5.71			6.01	20.01	20.0	3.23	4.33	8.32	5,32	2.65	-3.66	4.07	8.58	27.1	5.26	60.0	80.6	7,26	00.00	3.26	6.52		12.00		4 - 13	1,12	A.05	8.03	86.98	3,77	66.6	5.08	6.31	3.64		8.34	4.56	2,80	5.0%	3.70	5.74	2.05	3.03	9.50
		2														00	9 6		0	0									0.0	0 0			0 0		0	0 0	2 60	0						0	00		0
FRANSMITTER	POWER IN WATTS		50.0	30.	20.	10.	000	80.	10.	•	20.	20	50.	50.	20.	200	50.	50.	50.	50.	50,	200	10.	80.	50.	80.0	10.	10.	10.	50.	10.	50.	10.	50.	10.	200	50.	50.	20.	.01	50.	10.	10.	20.	30.	30.	10.
	LONGITUDE		10045001	751400#	77 236#	119 300#	10632001	11058001	971108*	984100W	1043630#	843000*	A921808	971400#	10416304	815100W	941800	10017001	801132*	841100#	855100×	337400	91 840#	864500W	622400*	AC 200%	651200#	674500W	#001 10	8 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	852717#	86 600*	40 447 W	8213041	89 300×	000 mm 00	40000000000000000000000000000000000000	614120#	835946#	#0000X	A95800#	864500#	801700%	881400#	92 200%	733228#	834800m
	LATITUDE		464700N	395300N	385042N	352558N	454800N	#11600N	455706N	#65500N	502325N	#62900N	UNDER JOHN	495400N	462200N	*12500N	200124	442300N	#02934N	3132004	333500N	NOOL SEE	303100N	333400N	362900N	385700N	35 200N	#1#700H	335700N	32.104.5N	311908N	335800N	332930K	3453484	302400N	3436198	521840N	502916N	3546412	312800N	SA ACON	322000N	254700N	304105N	323100N	4528#9N	31 SOON
	STATE OR PROVINCE		95	¥ a	**	**			OV.	ON.	SA	NO	200	N.A.	N. I	HO	2 2 3	20	PA	6.4	AL.	40	(*	Air.	12	××	12	11	35	6.8	N.	AL	0 1/2	SC	S	A P	200	FL	N.	S W	22	S×	1	AL.	3=	20	6.8
	CITY		BISMARCK	PHILADELPHIA	ARLINGTON	BAKERSFIELD	BILLINGS	FVANSTON	GRANDFORKS	JAMESTOWN.	REGINA	SAULT ST. MARIE	THUNDER BAT	*INTPE	BAKER	CLEVELAND	WINNEADOLTS	PIERRE	PITTSBURG	AL BANT	ANNISTON	ATHENS ATT ALTE	BATON ROUSE	BIRMINGHAM	BRISTOL	CHARLESTON	CHATTANDOGA	CHICAGO	COLUMBIA	COLUMBUS	DOTHAN	GADSDEN	GREENWOOD	GREER	GULFPORT	HUNTSVILLE	UACK SON	JACKSONVILLE	KNOXVILLE	LAUREL-MATTIE 18000	STHERE S	MERIDAN	MINNI	MOBILE	MONROE MONTGOMFRY	MONTREAL	MOULTRIE
	CALL		KANS					-Oxo		KAHS		VF.J7				KRKW	****	#8A.2	#HOW	KIDS	#103	A Inc				4.N.2	#8P3	*JZ3	2878	K106	N.JANS	K107	PLK 3	KU1,7	KLE W	200	KKGG	KIC6	WIVE		KID2			K162	ALSO WRY6	47.85	K105
	FREG.		131.69*0	131.6940	131.7000	131.7000	151.7000	131.7000	131.7000	131.7000	131.7000	131.7000	131.7000	131.7000	131.7060	131.7050	181.7060	131.7060	131.7060	131,7500	131.7500	131.7500	131.7500	131.7500	131.7500	131.7500	131.7500	131.7500	151.7500	131.7500	131.7500	131.7500	131.7500	131.7500	131.7500	151.7500	131.7500	131,7500	131.7500	131.7500	131.7500	151,7500	131.7500	131.7500	131.7500	131.7500	131,7500

S AVAILABLE COPY

(Page 4 of 9)

TABLE A-1

			,	*																																																
	Gard.	* C. N	1404																-	. 134.	0		01.									-138.	-1.40	-110.							-1 40.		-1.49.	-1.40		-130.				0.0	. 56.1-	
11E AT 180	2730	PONES IN US	GROCING																	-138		-1001										-100.	-138.	-145.			11.0				-114.		-138.	-136.		-136.	-138.				.001	7.87
SATELLITA	دادم	S. BAGLE	IN DES	-11.21	000	-20.30	-15.00	-18.28	-8.07	-12.23	-9.6-	-8.32	-13.33	-9.87	-10.56	-14.21	-11.52	-13.54	-10.88	16.22	2	13.60	20.00					-16.80	-10.12	-15.73	2.64	54.67	13,59	14.84	-6.36	-50.8			24.10		. 80	-17.27	12.74	12.46	-18.57	14.76	16.78	4.13	-10.12	13.98	100.00	
₩ 930	USA	× 00 ×	ACFT		-136				-139.		-130.	-139.		-139.	-139.				1	-138.	-139.	-138.			.001		-110				-138.	-137.	-138.	-158.	-130.		-138.		120.	-110.	-1.18		-138.	-138.		-138.	-138.	-139.		-130.	-138.	
ITE AT 155 DEG	37.30	PONTR IN DRY	GROUND		-138.				-138.		-145.			-138.					100000	-137.	-138.	-137.	-136.				-146				-137.	-143.	-137.	-144.	-138.								-1177.	-137.		-137.	-137.	-138.		-138.	-137	
SATELLITE		ANGLE		8.78			8.6.8	1.25	1.52	9.19	1.59	66.0	00.8	96.0	0.20	7.00	9.86	* . 95	-2.08	13.97	18.58	6.70	3.99	51.32		00.7	17:51	1.00	4.51		01.40	08.80	30.10	34.62	11.06	-2.56	33.84	2.00	11.00	100		. 52	. O. W.	25.92	58	30.10	34.72	20.57	7.92	12.32	33.25	0617-
			ACFT IN	-138.																			-135.												-135.			-138.				-116			-135.				-135.			-130.
SATELLITE AT NO DES .	713330	POWER IN DAN	GROUND AC	-1371																			-134			-153.			-154.						-134			-148.				1			-130.				-134			
LLITE A																			- 0	99	-			000									99											17			63					
SATE	2 12	ANGLE	1N 066	25.8	24.30	52.0		27.0	8.80	29.6	27.1	21.7	30.6	25.1	26.6	31.3	28.7	24.5	30.10	-2.3	7.7	-3.61	15.0	50,00	-77.5	28.0	31.95		27.1			-11.11	-1-	6.	16.83	32.0	1.72	36.	6.32	25.	10.	27.63		71.31	29.28	-3.	-2.63	9	21.	12.97	-6.18	
₩ 930	2000	1N 08*	ACFT					-1184								-139.			-136.							-135	-136.		-136.		.661-					-137.		-139.		130		-136	1001		-136.	-						-136
E AT 15	-30	POWER IN DRI	GENUND						1001							-145.			-129.							-134.	-135		-125.		-1001					-136.		-145.		-145.			-145.		-134.	-					-	-135
SATELLITE AT 15 DEG		ANGLE	1N DE 6	60.9	3.00	01.0	14.0	12.56	10.11			2.71	0.15	9.70	5.87	10.02	7.26	7.12	13.86	-21.31	-9.16	-20.46	-2.70	-21.18	-68.89	13.24	16.08	-12.74	10.43	26.0	11:10	70.00	110.52	-19.26	34	15.23	-18.49	13.68	-14.25	16.09	-7.20	-21.15	11.00	000	12.60	-21.05	-21.73	-10.97	3.81	-3.30	-22.35	15.08
	TRANSMITTER	IN WATTS		50.0	50.0	20.0	10.0	0.00	0.00	0.00			10.01	50.0	50.0	10.0	50.0																												20.05	20.0	50.0	50.0	51.0	50.0	50.0	20.0
	7.8	LONGITUDE		864100*	9:1730*	\$00\$106	735200	911900#	12384 / A	845200# 8541900	# 00 to 00 t	20000000	4421004	1018183	# 7 46 40 m	8316454	A6 1100w	833200*	74 035*	12158001	1055310#	1224700*	964600m	1225200#	1444800E	733900*	71 0000	111 930#	784130	875400	81 700	104601	10/0/00	11824004	W008186	73 600%	1172600*	801700*	112 100w	712345#	103 300×	1222030	774018#	1222030	10 4 10 M	1241400	1222300	1083200	875220#	97 006w	124 640#	72402Am
		LATITUDE		Ne 700N	313630N	295900N	1001 tos	282500N	452722N	342300N	2012004	205500	303000	20.50	111100	No about	102900W	421400N																													47.470 PM	454800N	425645N	475454N	404800N	2004314
		PROVINCE		2	SH	*7	-7	1	NO	5	,	2 3	2 4		2 -	140	i i	*	**		-	80	05	960	96	7	**	-	**	11	SC	0	ï		2 2	, AN	CA	FL	42	I a	05	3	× ×	*	30	200	4	3	*	Z	CA	40
		C117		A LECKSON	NATCHEZ	NE BORLEANS	NE B TORK	DRLANDO	01T0*A	OXFORD	PANAMA CITT	PASCABOULA	51 LOUIS	ALLANASSE.	10751	TUSCALDUSA VALODETA	1 Depart	YPSIL ANTI	, , , , , , , , , , , , , , , , , , ,	641961610	WILES CITY	PORTLAND	SIGUX FALLS	WEDFORD	AGANA	ALTONA	305TON	BOSEMAN	BUFFALO	CHICAGO	COLUMBIA	DENVER	HONOLULU	LAKEVIEW	MINNEADOLTS	STACK SAN	ONTARIO	OPA LOCKA	PHOENIX	PROVIDENCE	RAPID CITY	REDDING	ROCHESTER	SEATTLE	VANCOUVER	BINGHAMION	EUSENE CAN CRANCECCO	SAN FRANCISCO	GALLINGS C.DAHY	EAST GRAND FORKS	EUREKA	CADTERA
		CALL		3013				exr.				500	180		KKKB	0 C C C C C C C C C C C C C C C C C C C		4774	2 305						117	KETZ	KZ17	*OH7	KFVT	KPU7	#6x2	×1×2	#0x3	0 10 10	400		9000	91110	4109	#DA3	PAL 9	**13	*50*	*F07		-		1011	1212	5 N W W	NONE	VANA
		FREG		11 75.00	31.7500	31.7500	31.7500	31.7500	31.7500	31.7500	31.7500	31.7500	31.7500	31.7500	31.7500	31.7500	2000	11.7500	2000	11 7020	31.7920	31.7920	31.7920	131.7960	31.8000	131.8000	131.8000	31.8000	131.8000	131.8000	131.8000	131.8000	131.8000	131.8000	121.0000	2000	131.4060	131.8000	131.8000	131.8000	131.8000	131.8000	131.8000	131.8000	131.8000	131.8040	131.80*0	131.8040	131.8080	131.8080	131.8080	

TABLE A-1

(Page 5 of

	TVED	ACET									-1133,																-139,			*641-				
SAILLLIE AT 155 DEG # SAILLLIE AT 100 F F	PONES IN DAM	SROUND									-145139,																-145.			-145.				
	FLEV ANGLE	13 55	-13.10	-8.53	~6.80	18.39	-6.69	-18,32	-21.98	-13.38	16,19	-17,15	01:4-	-22.63	-21,01	-7.71	-20.84	-11,86	-55,45	-55.86	-20.39	-20.30	-50.10	-10.44	-22,17	~22,86	12.71	-5.38	-5.97	14.83	-7.63	-5.01	-8.62	-6.92
		ACFT		-139.	-139.	-139.	-139.				-138+		-139.			-139.											-138.	-139.	-130.	-138.	-139,	-139.	-139.	-1 30.
	POWER IN DB	GROUND		-138.	-138,	-138.	-138.				-144.		-132.			-138.											-144.	-138.	-138.	1744.	-145,	-138.	-138.	-1 80
	ELEV	14 050	7.46	12.51	13.64	13,22	14.63	.65	17:54	4.82	33.95	4.92	17.70	-4.73	-3.36	10.03	-3.40	7.54	54.45	-4.85	-2.04	-1.98	-1.75	82	E8.4-	-5.70	56.92	15.69	13.79	34.51	13.87	14.13	11.70	14.11
	IVED N DBK	ACET	-138.	-138.	-138.	-138.	-138.	-137,	-135,	-135.		-138.	-138.	-136.	-135.	-135.	-135.	-138.	-136.	1130	-136.	-136.	-136.	-137.	-134.	-134.		-138,	-139.		-138,	-137.	-138.	-138.
	POWER IN DRY	GROUND	-144.	-137.	-137.	-137.	-137.	-143.	-161-	-101-		. 1 ## .	+131.	-162.	-141.	-134.	-141.	-144.	-142.	-145.	-142.	-145,	-140.	-143.	-140.	-140.		-137.	-138.		-144.	-136.	-137.	-137.
	FLEV	N OE O	28.89	54.99	22.20	25.84	23,54	31,02	31.95	24.72	-2.34	36.40	21.63	32.93	30.04	18.60	30.25	25.45	32.95	33,30	31.75	31.59	31.47	31,41	31.26	31,52	-3.45	21.78	20.53	.8.	24.84	18.33	23,85	23.30
								-138.	-136.			-139,		-136.	-136.		-135.		-136.	-136.	-137.	-137.	-137.	-138.	-135.	-135.								
	POWER IN DRI	GROUND ACFT						-144.	-142.			-145.		-142.	-142.		-141.		-145.	-145.	-143.	-143.	-145.	-146.	-141.	-141.								
	ELEV ANGLE	IN CES	8.43	3.92	1.83	4.10	2.17	12,98	16.08	7.26	-21.29	13.77	21	16.91	14.96	1.20	14.54	6.43	16.77	17.23	14.80	14.67	14.40	13,95	16.00	16.62	-19.76	.70	.57	-19.29	3,25	-172	3.61	2,25
COLANGIANO	POWER IN KATTS		10.0	50.0	50.0	50.0	50,0	10.0	10.0	10.0	10.0	10.0	200.0	10.0	10.0	55.0	10.0	10.0	10.0	10.0	10.0	10.0	6.5	6.5	10.0	10.0	10.0	50.0	50.0	10.0	10.0	50.0	50.0	50.0
4.0	LATITUDE LONGITUDE IN WATTS		84.2500W	90 500×	921400W	901500×	921743W	77 200W	71 000W	832100W	1215600#	80 900×	951516w	701630*	721500#	911500W	721800W	854400×	703700*	70 400m	734600×	735200#	741000w	751754#	701600w	69 625#	1221810W	935313#	932200W	1182900#	#006 16	944320m	895800W	47 C CB
	LATITUDE		333900N	321900N	3444000	295900N	312348N	385100N	422200N	421300N	381600N	26 400N	293805N	413940N	425400N	435300N	433700N	381100N	412330N	#11517N	403800N	404600N	#0#200N	395212N	M33900N	44 330N	472709N	322723N	371420N	341200N	303200N	391825N	35 300N	TOTOTAL
	STATE OR		6.8	58	AR	4	47	VA	N.A	MI	4.5	73	TX	K M	T.V	1.4	EN.	KX	47	MA	N.X	×N.	2	P.A.	3×	WE	4.8	47	0	CA	14	OW	Z.	
	CITY		ATLANTA	JACKSON	LITTLE ROCK	NEW ORLEANS	AL EXANDRIA	ARLINGTON	BOSTON	DETROIT	FAIRFIELD	FT LAUDERDALE	HOUSTON	SINNEL	SAL SERVICE N	LA CROSSE	FBANON	LOUISVILLE	MARTHAS VINETARD	NANTUCKET	NE # YORK	NEW YORK	NEWARK	PHILADELPHIA	PORTLAND	ROCKALDN	SFATTLE	SHREVEDORT	SPRINGFIELD	SAON NEX	BATON ROUGE	KANSAS CITY	SIMONIA	T SUBJECT
	CALL		S. S	KKEG	KKG7	Ex 6.7	KKCR	KDV7	KUUS	640×	*OX#	*SXC*	*1.05	XCAS	8.069	87°	NO.	0 N J #	×CK5		KUV3		*PN2	# 598	m513	*CE6	*N66	SHAM	*B & 5	*700	VER.	X BA4	KIEG	
	- 2		5 4 4	44.5	588	5 8 8	200	200	800	200	2008	200	600	200	200	200	2005	2009	500	005	500	6500	2009	2005	1500	8500	1500	1500	1570	8500	1570	3570	15.70	25.20

		AT 180	POWER	GROUND						-140													-145.	-148.		-145.	-143.	-148.		-145.	-145.							-138.	-132.	
		SATELLITE AT 180	ELEV	N DE6	5.60	-6.83	-22.86	-20.10	-19.46	46.64	-18.32	-11.24	20.8	6.19	-10.12	-10.23	-3.20	-2,72	-2.13	14.50	90.41	-22.63	12.04	14.90	7.57	14:14	15.42	14.05	-17.29	16.78	14.48	9.00	-18.23	-20.91	6.26	-10.10	-19.86	13.72	12.21	-20.30
		₩ 930	_		-138.	-139.				-138.			-118	-138.			-139.	-139,	-139.	-138.	-139.		-138.	-138.	-138.	-118	-138.	-138.		-138.	-138.	-138.			-138.			-138.	-138.	
		E AT 155	POWER IN DRY	GROUND	-137.					-144.				-146.			-145.	-145.	-148.	-147.	-145.		-144.	-147.	-137.	-144	-144.	-147.		-144.	-144.	-137.			-137.			-137.	-131.	
		SATELLITE AT 155 DEG W	ELEV	IN DEG	21.19	10.07	-5.94	-2.87	10.	20.78	.65	9.41	22.16	22.83	8.22	8.10	18.37	19.26	19.41	26.48	17.66	-4.73	33.00	33.99	23.10	10.00	35.02	33.92		34.72	34.41	24.43	00.1	-2.97	22.11	7.65	-2.50	28.68	25.97	-1.96
		N 930	ETVED IN DBW	ACET	:	-135.	-134.	-134.	-137.	-138	-137.	-138.	-133		-136.	-136.	-139.	-139,	-139.	-138.	-138.	-136.	-136.			-130		:	135.				-138.	-136.	***	-135.	-133.			-136.
		SATELLITE AT NO DEG	POWER IN DBW	GROUND		-136	-127.	-127.	-136.	-137.	-143.	-146.	-168.		-135.	1100	-145.	-145.	-149.	-137.	-100.	-142.	-101-			-169.			- 100				-131.	-142.		-134	-132.			-129.
		SATELLIT	ELEV	IN DES	3.82	17.79	31.02	30.25	31.43	17.59	31.02	27.11	4.38	3.43	21.86	26.21	20.36	18.92	17.68	1.30	21.46	32.93	32.78		2.14	21.59	.25	1.72	27.62	-2.63	1.41	1.30	33.88	31.39	3.62	21.33	28.00	-3.42	-3.63	31.65
-	6	₩ 930	EIVED IN DBW	ACFT			-135.	-135.	-138.	-139.	-138.		-136.							-139.		-136.	-139.			-137.			-136.				1138	-136.			-134.			-137.
E A-1	go 9	SATELLITE AT 15 DEG	POWER IN DBN	BROUND			-128.	-128.	-137.	-138.	-144.		-153.						:	-138.		-142.	-132.			-120.			-135				-132.	-145.			-133.			-130.
TABLE	Page 6 of	SATELLI	ELEV	IN DE6	-12.75	55	16.46	13.71	13.98	11.24	12.98	6.52	10.00	-13.62	3.98	01.4	-1.24	-2.05	-2.86	13.77	08	16.91	111.45	-19.60	-14.58	14.80	-19.85	-18.51	11.09	-21.73	-18.87	-16.71	13.70	15.08	-13.23	3.78	12.84	-20.32	-19.54	14.69
		302	POWER IN WATTS		50.0	20.0	200.0	200.0	20.0	20.0	10.0	6.5	2002	6.5	50.0	0.00	10.0	10.0	8.0	20.0	10.0	10.0	200.0	0.5	50.0	200.00	10.01	5.0	20.0	10.0	10.0	20.0	200.0	10.0	20.0	20.00	50.0	0.00	200.0	200.0
-	3		LONGITUDE					73 900%																															12223491	735318#
6	round		CATITUDE		#72900N	**5100N	###800M	442800N	395215N	27583IN	385100N	333400N	#22200N	#55700N	#158#3N	N007414	303500N	325400N	342800N	26 420N	295800N	413940N	30291#N	353910N	465414N	10000 U	341200N	335700N	N046 64	373700N	334000N	473800N	38504#N	415500N	463627N	#3 62 PM	452734N	452720N	484226N	404128N
len For			STATE OR		L.	Z	¥	→ I > 2	A d	1	× ×	4	¥ ;	. 5	11	10	55	TX.	ě	1	S.Z.	**	10	55	T.W.		55	*5	2 1	25	45	×.	0 2		L X		200	OR	4.	r i
			CITY		GREAT FALLS	MINNEAPOLIS	BANGOR	BURLINGTON	PHILADELPHIA	TAMPA	ARLINGTON	BIRMINGHAM	BOSTON	BUTTE	CHICAGO	CHICAGO	COLLEGE STATION	DALLAS-FT. WORTH	DUNCAN	FT LAUDERDALE	HOUSTON	HANNIS	JACKSONVILLE	MC FARLAND	MISSOULA	NEW TORK	O CENTRO	RIVERSIDE	ROCHESTER	CAN FRANCISCO	SANTA ANA	SPOKANE	#ASHINGTON	WINDSOR LOCKS	HELENA	MADISON	MONTREAL	PORTLAND	SEATTLE	NEW YORK
6			CALL	p.				#J27											#Ex2	KT87	* SYS	XCA3	×166	X CS &	*OMS	KFK	KVV2	*055	KDLS									-		HE
C	Cored		FREG		131.89*0	131.8940	131.8945	131.8945	131.8945	131.8945	131.9000	131.9000	131.9000	131.9000	131.9000	131.9000	131-9000	131.9000	131.9000	131,9000	131.9000	131.9000	131.9000	131.9000	131.9000	131.9000	131-9000	131.9000	131.9000	131-4000	131.9000	131.9000	131.9000	131.9000	131.9060	131.9060	131.9060	131.9060	131.9060	131.9070

TABLE A-1

(Page 7 of 9)

N Sau	14 DBW	ACET																						-1.0	10614										
AT 190	POWER TH DRY	SROUND ADET																																	
SATELLITE AT 155 DEG W SATELLITE AT 180 NEG	ELEV ANGLE	200	-4.91	-19.21	-13,10	-18.00	-15.76	-17,36	-10.22	-4.31	-14.53	-12.62	-2.93	-6.17	2.70	-8.0%	-17.94	-19.46	68.4-	3.43	+7.71	-3.01	14.68	10.40	10.90	-20.51	-16.01	00.00	-20.14	-4.11	-2.63	-20.83	-5.07	-7.40	-15.38
N 930 S	IVED IN DBH	ACFT	-139.							-139.			-139.	-139.	-138.	-139.			-139.	-138.	-139.	-139.	-139.	-138.	-138.			-138.		-139.	-139.		-139.	-139.	
E AT 15	POWER IN DRI	BROUND	-148.							-145.			-147.	-147.	-144.	-145.			-148.	-147-	-145.	-145.	-158.	-147.	-147.			-144.		-147.	-148.		-148.	-147.	
SATELLIT	ELEV ANGLE	IN OF O	12.13	70	7.46	96.	4.29	1.85	8.17	17.51	3.87	6.52	18.05	10.29	21.45	10.13	.10	93	17.45	22.02	10.03	18.04	16.41	25.67	25.67	-1.88	5.64	22.40	-1.84	17.00	19.39	-2,95	12.22	10.28	3.03
× 030	IVED N	ACET	-134.	-136.	-138.	-137.	-138.	-137.	-136.	-138.	-136.	-137.	-139.	-136.	-138.	-136,	-135.	-136.	-138.		-135.	-136.	-138.			-136.	-136.		-136.	-139.	-139.	-135.	-134.	-134.	-136.
SATELLITE AT NO DER W	POWER IN DAN	SROUND	-143.	-145.	-137.	-146.	-147.	-143.	-142.	-144.	-142.	-143.	-146.	-144.	-144.	-145.	-102.	-142.	-147.		-141.	-144.	-157.			-135.	-145.		-145.	-146.	-148.	-141.	-143.	-145.	-135.
SATELLIT	FLEV	o and o	14.93	30.75	28.89	30.68	30.66	30.57	22.06	21,88	26.21	25.72	19.16	20.22	10.52	19,72	28.76	32,26	22-11	9,65	18,50	21.68	21.11	-1.43	-1.43	31.54	28:12	8.93	31.41	20.55	20.44	31.18	15.47	18.21	27.05
× 930		ACFT		-137.		-138.	-139.	-139.									-136.	-217.								-137.	-108.		-137.			-136.			
E AT 15	POWER IN DBM	6ROUND		-143.		-147.	-148.	-145.									-143.	-143.								-136.	-144.		-143.			-142.			
SATELLITE AT 15 DEG W	ELEV	IN CE	-1.98	13,57	8.43	12.63	10.95	12.11	4.12	+01	8.60	7.06	-1.82	2.07	-8.48	1.84	11.96	13.92	.16	-9,25	1.20	32	.01	-17.92	-17.92	14.59	10.31	-9.86	14.49	57	-1.59	14.94	-1.68	.85	9.50
	POWER IN WATTS		5.0	10.0	50.0	5.0	5.0	10.0	10.0	10.0	10.0	10.0	6.5	6.5	10.0	10.0	0.0	10.0	5.0	5.0	10.0	10.0	in.	5.0	5.0	50.0	10.0	10.0	10.0	6.5	5.0	10.0	5.0	6.5	50.0
1	LONGITUDE		952335.	7526451	842600*	772700×	805600 w	782700%	874630×	95 125w	815030W	844000m	965180W	9035009	1045300%	904300×	765400W	751230w	945200w	1055500W	911500w	952712	944300×	1191900#	1191900W	74 000 W	795600w	10644238	74 400%	952400W	965500N	7204004	95 500#	914200*	80#00m08
	LATITUDE		445220N	403820N	333922N	385700N	351300N	38 800N	414600N	293326N	412400N	39 300N	325100N	#13600N	394600N	422400N	421000N	40 620N	291600N	40 SOON	435300N	29 210N	322300N	#71200N	#71200N	N005+0*	402100N	402726N	405100N	\$22100N	285100N	MO00 CH	45 700N	N005 ##	411600N
	STATE OR PROVINCE		2	PA	6.4	* >	NC	V.A	II	Tx	HO	K.Y.	Tx	1.4	000	1.4	N.Y.	P.A.	TX	000		TX	Tx	¥.4	N.A.	L'A	PA	00	Z	1.4	1.4	***	Z	NN	HO
	2117		AL F. YANDRIA	ALLENTOWN	ATLANTA	CHANTILLY	CHARLOTTE	CHARLOTTESVILLE	CHICAGO	CLEAR LAKE CITY	CLEVELAND	CONTNETON	DALLAS	CAVENDORT	DENVER	DUBLISHE	FLMIRA	ST MASHINGTON	GALVELSTON	GRANEY	LACROSSE	LAKE JACKSON	LONGVIEW	MOSES LAKE	MOSES LAKE	NEW YORK	PITTSBURG	STEAMBOAT SPRINGS	TETERRORO	TYLER	VICTORIA	*ECTETEIN	#ILLMAR	#1 NON #	TOUNGSTOWN
	CALL		5,46	8448	KIES	010	1000	*N0.7	4413	x7.13	* SR.	2000	KHAN	0000	***	K.344.3	*NKS	4 25 4	× DE 8	WNX7	KKKK	K.807	KE G.S.		KHP3	KIND	*		-	d July			6 13		× 9×2
	FREG IN MHZ		111.9500	131.9500	131.9500	141.0500	181.9500	111.9500	171.9500	131.9500	131.9400	1 11 . 05.00	181.0500	141.05600	131.9500	181.9500	131.0500	181.0500	131.9500	131.9500	141.9500	131.9500	131.9500	131.9500	131,9500	131.9500	131.9500	131.9500	131.9600	1 81 . 0500	1X1.9500	111 0500	131.9500	131.9500	131.9500

BEST AVAILABLE COPY

TABLE A-1 (Page 8 of 9)

																																												•
* 530 0	POWER IN DRY		ACE T					-133.																													-110.							
E AT 18	DO NEB		GROUND					-130.																													-108.							
M SATELLITE AT 180 PES	FLEV	1" DEG		-19.21	-8.56	-7.66	-13.38	16.36	-9.63	4.30	-13.11	-11.51	-25.42	-15.01	-16.17	-11.17	-7.57	-5.76	-18.20	-8.63	-9.18	-9.30	-10.34	-10.12	-19.05	-13.90	-2.93	-2.78	2.70	-13.38	-13.24	70.7.		20111	17.15	16.51	14.75	-7.96	8.78	-16.22	-3.97	-11.38	-6.78	
¥ 930 €	IVED IN		ACET		-139.	-139.		-132.		-138.							-139.	-139.		-139.	-139.	-139.					-130.	-139.	-138.			-134.				-130.	-138.	-139.	-138.		-139.		-139.	
F AT 15	POWER IN DRW		GROUND		-147.	-138.		-138.		-144.							-138.	-138.		-145.	-140.	-147.					-145.	-145.	-144.			-128.				-138.	-144.	-145.	-144.		-142.		-138.	
SATELLITE AT 155 DEG	ELEV	IN DEG		64	10.70	14.04	7.54	20.60	8.00	23.15	7.45	6.23	-8.72	6.19	4.55	7.10	14.37	15.97	.39	10.80	10.39	10.06	8.03	8.22	0 +	66.4	18.05	18.19	21.45	4.82	4.95	16.1	***	000		17.18	14.58	11.47	25.53	4.55	17.73	7.60	14.92	10.22
DE6 W			ACFT	-136.	-138.	-138.	-138.		-134.		-138.	-135.	-134.	-135.	-138.	-135.	-138.	-138.	-136.	-138.	-138.	-138.	-136.	-136.	-137.	-136.	-139.	-139.	-138.	-135.	-135.	-134.	-138.			-138.		-138.		-138.	-138.	-137.	-138.	
E AT 40	POWER IN DRW		GROUND	-142.	-146.	-137.	-147.		-140.		-104.	-143.	-140.	-143.	-147.	-144	-137.	-137.	-145.	-104.	-147.	-146.	-100	-142.	-136.	-145.	-145.	-145.	-144.	-143.	-134.	-156.				-137		-148.		-147.	-141.	-143.	-137.	
SATELLITE AT 40 DE6	FLEV	1N 0E6		30.74	21.95	25.23	29.89	-17.75	20.19	9.22	28,90	22.11	32.82	23.42	32.45	22.75	25.53	23.26	30.12	23.13	53.09	22.84	22.13	21.86	30.98	56.40	19.16	18.90	10.52	24.72	24.59	15.71	30.01	13 10	14.40	22.18	1.06	21.64	2.66	32.62	21,34	24.22	24.30	2017
DE6 *	IVED IN DBW		ACET	-137.									-134.		-139.				-137.						-138.															-139.				
SATELLITE AT 15 DE6	POWER IN DB		6ROUND	-143.									-140.		-148.				-143.					-	-137.									-186						-148.				
SATELLIT	ELEV	IN DEG		13.56	2.93	3.40	96.8	-27.62	3.07	46.6-	8.45	5.04	19.06	5.85	11.82	2.03	3.43	1.45	12.59	3.59	3.74	3.74	4.23	3.98	13.50	8.20	-1.85	-1.98	-8.48	7.26	7.12	-3.88	9.99		12.77	30.	-19.17	2.41	-15.10	11.91	39	2.67	5.49	
STATE OF STATE OF	POWER			10.0	6.5	50.0	5.0	10.0	10.0	10.0	10.0	6.5	10.0	6.5	2.0	6.5	20.0	20.0	10.0	10.0	5.0	6.5	6.5	10.0	20.0	10.0	10.0	10.0	10.0	6.5	20.0	20.00	0.0			20.0	10.0	10.0	10.0	5.0	20.0	10.0	20.0	
•	LONGITUDE			752600w	90 300W	91 618w	8411004	1495900	983110w	1064913#	842510W	855100w	#006**9	851500W	804300m	862600w	9112054	932020	765600W	8934201	891500#	89 500m	873600W	875421#	755130W	825230W	965100	97 200W	1045300W	832100m	833200M	100000	825900W	100000	1000	9444	1181635w	9050624	1141830w	804150#	952606#	861700W	92 838W	-
	LATITUDE			M00650*	385300N	294138N	32 700N	611000N	441540N	39 90 BN	333925N	435300N	1008 ***	421900N	322830N	45 SOON	284105N	294640N	#01##0W	371338N	374 700N	383100N	415100N	415843N	395845N	M000 0%	325100N	325400N	394600N	421300N	451400N	3746000	3234004	11,300	1004	20103EN	335123N	362018N	433030N	321320N	30 313N	39420BN	294710N	1000000
	STATE OR			¥ d	11	5	6.4	AK		00	6.4	ï	3	7	SC		5	5	¥ d	9	11.	11	11	11	¥ d	¥	X	X	00			2	6A	2 4	40	12	. 5	KS	ID	200	1×	Z.	5.	11
	2117			ALLENTOWN	ALTON	AMELIA	AMERICUS	ANCHORAGE	APPLETON	ASPEN	ATLANTA	BALDWIN	BANGOR	BATTLE CREEK	BEAUFORT	BENTON HARBOR	BLK 1540 SHIP SHOAL	CAMERON	CAMP HILL	CAPE GIRARDEAU	CARBONDALE	CENTRALIA	CHICAGO	CHICAGO	COATESVILLE	COLUMBUS	DALLAS	DALLAS-FT. WORTH	DENVER	DETROIT	DETROIT	DODGE CITY	DUBLIN	ELANAR!	CT LANGORNE	GAL VESTON	GERDINA	GREAT SEND	HAILEY	HILTON HEAD ISLAND	HOUSTON	INDIANAPOLIS	INTERCOASTAL CITY	JACKSONFILLE
	CALL			KP92	1527	WK 95	WFA3	#E16	KTTS	KYC9	KNAS	#RP+	KPS7	#PI6	KW63	*RP4	WK67	#KT4	*8F7	KPY6	WHYZ	K RR4	#NC5	KL66	#ONB	6788	# OH5	xx.55	KWJ8	#PF3	KRY7	KAH9	***	404	X L	220	*0.4	8 A 6 9	K Bu 2	× TR4	#8P2	8 X0 8	35.	2
	FREG.			132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.3000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	135.000	132.000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000	132.0000

TABLE A-1

(Page 9 of 9)

	OULT INSTRUCT
	STATE OR PROVINCE LATITUDE LONGITUDE IN WATTS
IN DE	
TX 31 500M 974100W 10.0	TX 31 500M 974100W 10.0
10.00 MOSSON 86 MOSSON 10.0	0.01 NOOP 90 NOONO NO
0.05 MARGINE MORCHUE A.	0.05 MARGINE MORCHUE A.
NV 36 500N 115 900N 6.5	NV 36 500N 115 900N 6.5
NV 361215N 1151150% 5.0	NV 361215N 1151150N 5.0
LA 291330N 901415% 50.0	LA 291330N 901415% 50.0
D*O BOD DTT NOOFIE	D*O BOD TT NOOFICE TO
0.00	0.00 2000000000000000000000000000000000
O'CI ACCOUNCE NOOFINE	O'CL MODELLE MODELLE NOT THE TOTAL T
MN 445300N 931300* 10.0	MM 445300N 931300* 10.0
#1 #84600N 894000W 10.0	#1 #84600N 894000W 10.0
272930N 971500# 50.0	TX 272930N 971600# 50.0
NAS NAS 295637N 901109W 50.0	LA 295637N 901109W 50.0
X 25500 X	0.00 1002001 N000000 NN
0.00 101 101 101 101 101 101 101 101 101	STATE SOUTH STATE OF THE STATE
4 NOKAGON 751400W 11.0	4 NOKAGON 751400W 11.0
453500M 1223600W 5.0	OR 453500M 1223600W 5.0
1L 395700W 911200W 6.5	1L 395700W 911200W 6.5
#A 473100N 1223600W 10.0	#A 473100N 1223600W 10.0
NO 365300N 893800N 5.0	MO 365300N 893800N 5.0
TY 122400N 962300N 2+0	TY 122400N 962300N 2+0
TN 36 000M 863100W 10.0	TN 36 000M 863100W 10.0
15LANG LA 285259N 915545H 50.0	15LANG LA 285259N 915545H 50.0
0.00 moores Noores 71	0.00 moores Noores 71
N. N. NOCTOCO MOCONER	NOTICE MECCAMEN
7 274500N A28100W 10.0	7 274500N A28100W 10.0
40 6A 31 900N 812300W 10.0	40 6A 31 900N 812300W 10.0
10 6A 322900N 814500W 5.5	6A 322900N 814500W 5.5
OH 41 030H 812130W 20.0	OH 41 030H 812130W 20.0
*A 87 900M 1222900W 10.0	*A 87 900M 1222900W 10.0
#00100N 7# #00#	20 #00# #4 NO0190# 72
177E 64 305400N 833300N 6.5	177E 64 305400N 833300N 6.5
6A 312600N 832900W 5.0	6A 312600N 832900W 5.0
LA 291550N 892185W 50.0	LA 291550N 892185W 50.0
LA 283434N 922638N 100.0	LA 283434N 922638N 100.0
VERMILION BLK 76 291719N 922320N 10.0	VERMILION BLK 76 291719N 922320N 10.0
6A 311500N 822400N 10.0	MAYCROSS GA SILSOON B22400N 10.0
MEST CAMERON BK 543 LA 282040N 822400V 10.0	MEST CAMERON BK 543 LA 282040N 822400V 10.0
LAINS NY 81 421N 734225W 10.0	The same of the sa
2008000 200000	AINS NT 61 62IN 1342ESB

APPENDIX B

TRANSMITTERS IN THE US AND CANADA IN THE 125.4 to 126.0 MHz BAND

TABLE B-1 is a list of all the ground-based transmitters in the US and Canada in the 125.4 to 126.0 MHz band. (They all use omni-directional antennas.) The column headings are described below:

FREQUENCY IN MHz - gives the exact carrier frequency of each transmitter. The transmitters are listed by frequency.

STATE OR PROVINCE - ground station location.

CITY - ground station location.

LATITUDE - ground station location.

LONGITUDE - ground station location.

TRANSMITTER POWER IN WATTS - ground station transmitter power.

TABLE B-1
TRANSMITTERS IN THE US AND CANADA IN THE 125.4 TO 126.0 MHz BAND (Page 1 of 12)

Transmitter Power In Watts	10	20	10	10	10	10	10	10	10	20	50	50	50	10	50	50	50	20
Longitude	110 49 13W	094 22 XXW	N2 119 51 27W	121 33 52W	080 06 51W	081 59 27W	087 55 XXW	096 06 43W	082 54 19W	075 21 XXW	086 47 XXW	097 30 34W	097 13 50W	101 49 35W	111 57 31W	122 28 XXW	97 14 00W	79 38 00W
Latitude	33 16 56N	35 19 XXN	34 25 47N	39 06 11N	26 14 36N	33 22 30N	41 59 XXN	41 11 46N	39 59 56N	40 22 XXN	36 21 XXN	27 45 53N	31 37 13N	35 10 15N	40 46 27N	47 09 XXN	49 54 00N	43 41 00N
City	Globe	Ft. Smith	Santbrbr	Marysvll	Pompnbch	Augusta	Chicago	Millard	Columbus	Flinth11	Nashv11	Crpschrs	Waco	Amarillo	Saltl Cy	Тасоша	Winnipg	Toronto
State or Province	AZ	AR	CA	CA	FL	GA	IL	NE	НО	PA	N.	TX	TX	TX	TI	WA	Mani toba	Ontario
Frequency In MHz	125.4	125.4	125.4	125.4	125.4	125.4	125.4	125.4	125.4	125.4	125.4	125.4	125.4	125.4	125.4	125.4	125.4	125.4

TABLE B-1

(Page 2 of 12)

Frequency In MHz	State or Province	City	Latitude	Longitude	Transmitter Power In Watts
125.45	CA	San Diego	32 49 15N	116 59 15W	10
125.45	FL	Jcksnv11	30 28 32N	081 39 10W	50
125.45	11	Quincy	39 56 39N	091 08 56W	50
125.45	MI	Pontiac	42 39 44N	083 25 22W	10
125.45	NY	Albany	42 44 32N	073 49 21W	10
125.5	AL	Montgmry	32 22 XXN	086 24 XXW	20
125.5	CA	Mtn View	37 19 08N	122 08 45W	20
125.5	CA	Ontaria	34 03 15N	117 35 15W	10
125.5	CA	Sacramnt	38 33 14N	121 16 09W	20
125.5	CA	Oakland	37 31 44N	122 25 35W	20
125.5	CA	Oakland	37 32 33N	122 00 53W	20
125.5	FL	Petrsbrg	27 54 23N	082 41 27W	50
125.5	ID	Boise	43 30 34N	116 14 07W	50
125.5	KS	Hutchnsn	38 03 57N	097 52 06W	10
125.5	LA	N Orleans	29 59 XXN	090 15 XXW	50
125.5	MI	Bntnhrbr	42 07 30N	086 25 37W	10
125.5	W.	Darwin	45 05 15N	094 27 12W	50
125.5	WO	St. Louis	38 48 52N	090 23 09W	50

TABLE B-1

Frequency In MHz	State or Province	City	Latitude	Longitude	Transmitter Power In Watts
125.5	MO	St. Louis	38 42 04N	090 23 26W	20
125.5	NV	Battlmtn	40 24 11N	116 52 02W	20
125.5	ĹN	Newark	40 41 XXN	074 09 XXW	50
125.5	NY	Buffalo	42 56 11N	078 44 39W	50
125.5	НО	Akron	40 S4 45N	081 26 13W	50
125.5	CZ	Balboa	08 55 36N	079 37 23W	1000
125.5	SC	Greenv11	34 39 XXN	082 07 XXW	50
125.5	XT	Austin	30 17 30N	097 41 55W	10
125.5	VA	Chant11y	38 56 31N	077 25 42W	50
125.5	British Columbia	Сотох	49 43 00N	124 54 00W	50
125.5	Ontario	Sudbury	46 38 XXN	080 46 XXW	50
125.5	Quebec	Roberval	48 31 00N	072 17 00W	50
125.55	FL	Orlando	28 32 42N	081 20 29W	50
125.55	NY	Wayland	42 30 XXN	077 37 XXW	. 05
125.55	НО	Marietta	39 27 44N	081 21 45W	50
125.55	WI	Greenbay	44 20 26N	087 56 48W	20

TABLE B-1
(Page 4 of 12)

Frequency	State or	, i	12+1140	I ongi tudo	Transmitter
7111	LIOVING	CTC)	Partrage	nough care	TOWN THE WALLS
125.6	AL	Huntsv11	34 38 55N	086 46 21W	50
125.6	AZ	Phoenix	33 26 10N	111 59 45W	10
125.6	AR	Littlrck	34 43 48N	092 13 23W	50
125.6	CA	Snfrncsc	37 37 14N	122 21 52W	10
125.6	CA	Torrance	33 48 06N	118 20 21W	10
125.6	00	Denver	39 45 18N	104 53 54W	10
125.6	FL	Miami	25 47 58N	080 21 07W	50
125.6	IL	Lwrncv11	38 45 XXN	087 36 XXW	10
125.6	MI	Detroit	42 13 XXN	083 22 XXW	10
125.6	IM	Carleton	42 00 12N	083 21 44W	10
125.6	QN	Jamestwn	47 06 24N	099 18 40W	50
125.6	NV	Las Vegas	36 04 45N	115 08 56W	10
125.6	Ν̈́	Shipbttm	39 47 XXN	074 21 XXW	50
125.6	NM	Tucumcar	34 56 15N	103 23 30W	10
125.6	MN	Mesarica	35 14 17N	104 12 14W	20
125.6	NY	NY Cy	40 39 XXN	073 46 XXW	50
125.6	OK	Okla Cy	35 23 16N	WSE 35 Y60	20

TABLE B-1

(Page 5 of 12)

Frequency	State or	3	9	opui tudo	Transmitter
7114 1117	LIOVINCE	CLC)	opport and	congrada.	cannot to to to
125.6	TN	Bristol	36 28 30N	082 24 15W	10
125.6	TX	Houston	29 33 27N	W12 80 21W	50
125.6	III	Saltl Cy	40 46 27N	111 57 31W	10
125.6	WA	Ft. Lawton	47 39 22N	122 24 43W	50
125.6	WA	Ft. Lawton	47 39 28N	122 24 43W	20
125.6	WA	Auburn	47 17 16N	122 11 12W	20
125.6	WI	Dells	43 33 03N	089 45 49W	10
125.6	Montreal	Quebec	45 27 00N	073 45 00W	50
125.65	CT	Wndsrlks	41 56 22N	072 40 31W	10
125.65	FA	Gainsvll	29 41 27N	082 16 28W	10
125.65	KY	N Hope	37 37 54N	085 40 33W	50
125.65	MD	Cp Spr	38 48 XXN	076 53 XXW	50
125.65	MI	Gr Rapids	42 52 31N	085 31 46W	10
125.7	AK	Homer	59 40 31N	151 35 08W	C.
125.7	AL	Mobile	30 40 54N	088 14 33W	50
125.7	AK	Fire I	N8E 60 19	150 19 29W	50
125.7	AR	Texarkan	33 27 24N	093 59 45W	50

TABLE B-1

(Page 6 of 12)

Frequency In MHz	State or Province	City	Latitude	Longitude	Transmitter Power In Watts
125.7	CA	Sacramnt	38 41 59N	121 35 33W	10
125.7	00	Colo Spr	38 49 23N	104 42 07W	10
125.7	GA	Atlanta	33 39 XXN	084 26 XXW	50
125.7	ID	Lewiston	46 22 28N	117 00 53W	10
125.7	II	Chicago	41 59 XXN	087 55 XXW	10
125.7	KS	Wichita	37 40 XXN	097 26 XXW	50
125.7	MN	Minnepls	45 03 37N	093 20 39W	10
125.7	NE	Omaha	41 07 23N	WS0 55 03W	10
125.7	NE	Lincoln	40 50 46N	096 46 34W	100
125.7	NY	NY Cy	40 38 46N	073 46 55W	50
125.7	НО	Chardon	41 31 01N	081 09 48W	50
125.7	TX	Sananton	29 32 18N	098 28 01W	50
125.7	TI	Saltl Cy	40 46 43N	111 57 21W	10
125.7	VA	Norfolk	36 53 27N	076 12 35W	50
125.7	VA	Nwprtnws	37 07 58N	076 29 36W	10
125.7	AK	Talkeetn	62 19 53N	150 05 56W	50
125.7	Alberta	Edmonton	53 19 XXN	113 35 XXW	50

TABLE B-1

(Page 7 of 12)

Power In Watts Transmitter 50 50 10 10 50 50 10 20 20 10 10 10 50 41W 38W XXW 23W 32W 50W 35W 58W 34W 10W 51W 28W 29W 04W 04W SOW 49W Longi tude 30 20 10 44 00 03 10 39 05 31 07 980 103 122 104 104 104 091 094 077 34N 55N 55N 00N 52N 51N 44N 5 2N 33N 44N 111 31N 30N XXX 06N 5 0N 05N Latitude 99 44 36 39 05 17 20 30 47 35 04 44 32 35 47 41 37 39 40 31 37 37 45 34 33 34 37 37 Kansas Cy Saddlepk Trinidad Teterbor Buenavst Killaloe Sanpedro Palmdale La Junta City Eastnv11 Montreal So Bend Narchez Bedford Hayward Parker Denver Miami Ontario Province Quebec State or S VA Z WS 9 VA CA CA CA CA 00 9 00 Frequency 125.75 125.75 125.75 125.75 125.75 125.75 125.75 In MHz 125.8 125.7 125.8 125.8 125.8 125.8 125.8 125.8 125.8 125.8

TABLE B-1 (Page 8 of 12)

	-		The state of the s		
Frequency In MHz	State or Province	City	Latitude	Longitude	Transmitter Power In Watts
125.8	CN	Woodstck	41 57 XXN	072 13 XXW	50
125.8	FA	Татра	28 29 XXN	082 23 XXW	50
125.8	ID	Ashton	44 33 41N	111 26 36W	20
125.8	1I	Peoria	40 40 XXN	089 41 XXW	10
125.8	KY	Owensbor	37 44 46N	087 09 40W	10
125.8	MA	Winthrop	42 23 21N	070 58 07W	20
125.8	N	Duluth	46 50 16N	092 11 23W	10
125.8	OW.	Springfield	37 14 38N	093 22 52W	10
125.8	MT	Billings	45 48 37N	108 35 40W	10
125.8	NC	Ashevil1	35 26 04N	082 32 04W	10
125.8	N	Reno	39 31 53N	119 39 18W	10
125.8	妥	Nashua	42 44 10N	071 28 52W	20
125.8	NY	Saratoga	43 00 37N	073 40 57W	20
125.8	Ю	Dayton	39 54 01N	084 15 01W	10
125.8	OR	Horton	44 17 01N	123 35 09W	50
125.8	OR	Laurlmtn	44 55 24N	123 34 23W	20
125.8	PR	San Juan	18 26 24N	065 59 53W	10
125.8	SD	Sioxflls	43 36 XXN	096 44 XXW	10

IIT RESEARCH INST ANNAPOLIS MD F/6 17/2.1 EMC ANALYSIS OF THE PROPOSED AEROSAT VHF SUBSYSTEM WITH CURRENT--ETC(U) MAY 77 F M SLAY F19628-76-C-0017 AD-A044 004 UNCLASSIFIED ECAC-PR-75-071-VOL-1 NL 2012 AD 44004 END DATE 10 -77 DDC

TABLE B-1 (Page 9 of 12)

Frequency In MHz	State or Province	City	Latitude	Longitude	Transmitter Power In Watts
125.8	NI	Memphis	35 04 XXN	MXX 65 680	95
125.8	TX	Ft. Worth	32 49 17N	097 02 06W	20
125.8	TX	Dallas	32 51 18N	096 50 49W	50
125.8	TX	Lubbock	33 40 06N	101 49 54W	50
125.8	TX	Grapevin	32 54 45N	097 05 28W	50
125.8	WA	Spokane	47 36 10N	117 39 30W	20
125.8	WY	Sheridan	44 36 44N	106 45 27W	50
125.8	MY	Lovell	44 49 00N	107 54 06W	20
125.8	Ontario	St Caths	43 11 00N	79 10 00W	90
125.85	FL	Jcksnvll	30 30 08N	081 41 29W	10
125.85	IA	Desmoins	41 31 49N	093 38 42W	10
125.85	N	Newark	40 41 XXN	074 09 XXW	50
125.85	НО	Cleveland	41 29 29N	081 43 33W	10
125.85	Ĭ.	Graham	35 50 30N	087 25 56W	50
125.85	VA	So Boston	36 40 30N	079 00 53W	10
125.85	WI	Milwauke	42 57 00N	087 54 02W	10
125.9	AZ	Dvsmnthn	32 10 15N	110 52 37W	10
125.9	CA	SanCarls	37 30 38N	122 14 27W	10

TABLE B-1

(Page 10 of 12)

Frequency In Miz	State or Province	City	Latitude	Longitude	Transmitter Power In Watts
125.9	FL	Opalocka	25 54 40N	080 16 50W	50
125.9	В	Atlanta	33 22 SON	084 18 00W	50
125.9	KS	Hutchnsn	37 55 24N	097 54 14W	20
125.9	KS	Hutchnsn	37 59 56N	W00 78 760	90
125.9	QW	Baltimor	39 11 XXN	076 41 XXW	90
125.9	MI	Lansing	42 46 21N	084 35 46W	10
125.9	M	Farmngton	44 38 XXN	MXX 60 £60	20
125.9	W	Farmngton	44 38 15N	W20 60 860	20
125.9	WS	Jackson	32 19 XXN	090 04 XXW	20
125.9	WO	Stcharls	38 45 39N	090 31 13W	20
125.9	N	Las Vegas	36 05 XXN	115 10 XXW	20
125.9	PA	Pttsbrgh	40 30 XXN	080 12 XXW	20
125.9	XT	Temple	31 09 10N	097 24 23W	50
125.9	MA	Seattle	47 28 18N	122 18 16W	20
125.9	WA	Mcchord	47 08 44N	122 28 05W	50
125.9	MV	Wheeling	40 10 49N	080 38 24W	10
125.9	Quebec	Val D'or	48 03 00N	077 47 00W	90
125.9	Newfoundland	Gander	48 57 00N	54 34 00W	20

TABLE B-1 (Page 11 of 12)

Transmitter Power In Watts	W 50	W 50	10 NI	.W 10	9W S0	10 W	. 10 Me	IN 20	1W 20	3W 10	8W S0	N9 20	0S MC	SW 50	3W 10	W 10	
Longi tude	104 43 30W	093 32 50W	074 29 44W	077 40 01W	082 54 19W	W61 61 770	122 13 29W	118 20 04W	118 05 04W	118 22 38W	075 36 58W	082 31 35W	084 56 30W	156 16 03W	090 03 28W	094 44 11W	11000
Latitude	39 44 12N	32 38 30N	40 12 08N	43 07 35N	39 59 56N	37 30 27N	37 42 46N	33 44 44N	34 36 11N	33 57 44N	39 40 56N	27 57 51N	32 31 05N	20 42 31N	38 53 23N	38 51 15N	
City	Denver	Shrevprt	Robbnsv11	Rochestr	Columbus	Richmond	Oakland	San Pedro	Palmdale	Los Angeles	Wilmngtn	Татра	Columbus	Haleakal	Alton	Olathe	
State or Province	00	LA	N	ΝΥ	НО	VA	S.	CA	CA	CA	DE	FL	GA	HI	11	KS	•
Frequency In MHz	125.95	125.95	125.95	125.95	125.95	125.95	126	126	126	126	126	126	126	126	126	126	701

TABLE B-1
(Page 12 of 12)

Frequency	State or				Transmitter
In Miz	Province	City	Latitude	Longitude	Power In Watts
126	OR	Newberg	45 21 12N	122 58 37W	10
126	SC	Charlstn	32 53 45N	080 02 15W	50
126	TX	Ft Worth	32 56 35N	097 21 06W	20
126	XT	El Paso	31 51 SIN	106 29 24W	20
126	XT	El Paso	31 40 S3N	106 11 50W	20
126	Ħ	0gden	41 11 46N	112 01 16W	10
126	VA	Roanoke	37 18 33N	M9E 60 080	20
126	WI	Wausau	45 01 34N	089 35 18W	20
126	Alberta	Edmntn'	53 19 XXN	113 35 XXW	20
126	Ontario	Sarnia	42 S4 00N	082 18 00W	20
126	Quebec	Mt Joli	48 36 00N	068 12 00W	1000
126	Onepec	Sept 11s	50 13 00N	066 16 00W	50

APPENDIX C

COVERAGE AREAS OF EXTENDED RANGE VHF STATIONS IN THE US, CANADA, AND THE NORTH ATLANTIC REGION

FIGURES C-1 through C-6 show the coverage areas of the extended range VHF stations in the US and Canada. FIGURE C-7 shows the coverage areas of the extended range VHF stations in the North Atlantic Region.

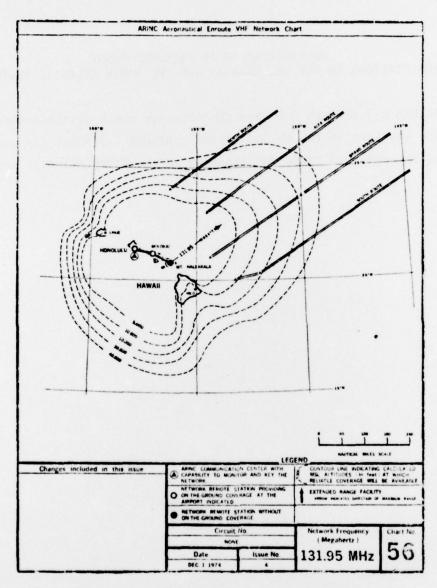


FIGURE C-1. EXTENDED RANGE VHF TRANSMITTERS IN THE US AND CANADA, HAWAIIAN ISLANDS.

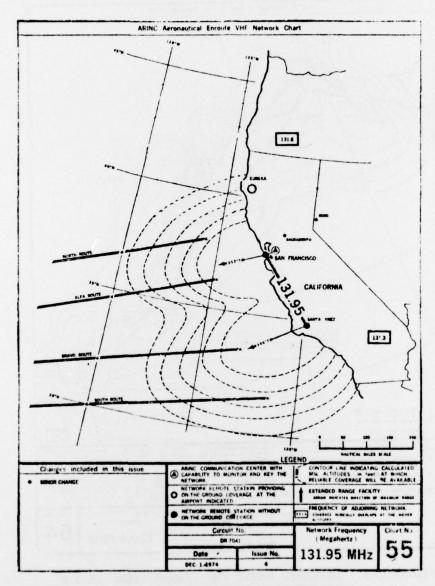


FIGURE C-2. EXTENDED RANGE VHF TRANSMITTERS IN THE US AND CANADA, WEST COAST US.

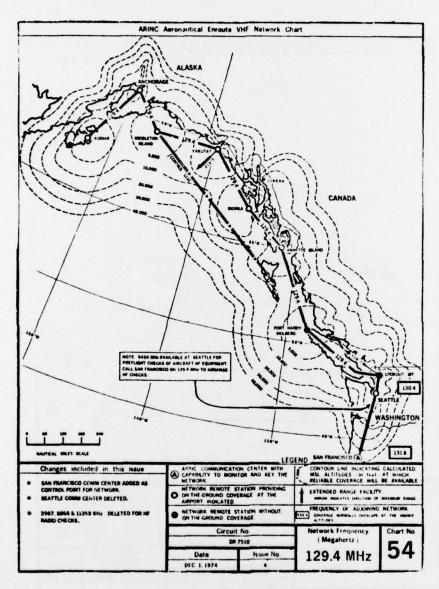


FIGURE C-3. EXTENDED RANGE VHF TRANSMITTERS IN THE US AND CANADA, CANADA AND ALASKA.

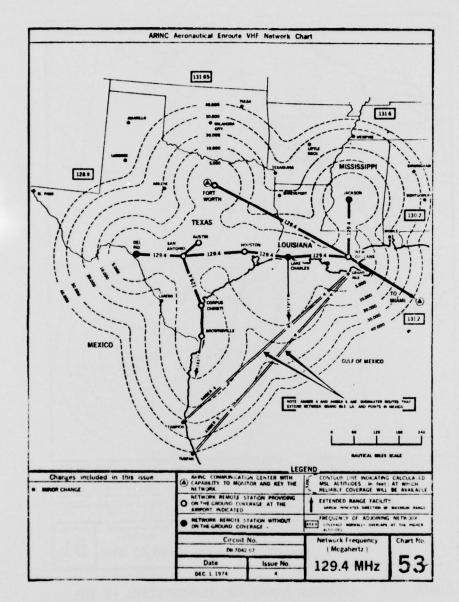


FIGURE C-4. EXTENDED RANGE VHF TRANSMITTERS IN THE US AND CANADA, GULF COAST.

BEST AVAILABLE COPY

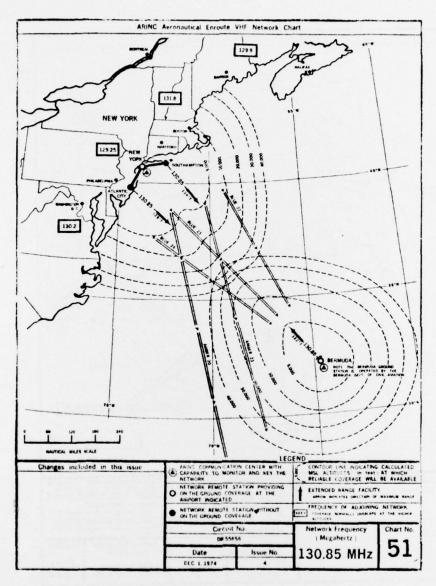


FIGURE C-5. EXTENDED RANGE VHF TRANSMITTERS IN THE US AND CANADA, EAST COAST, US.

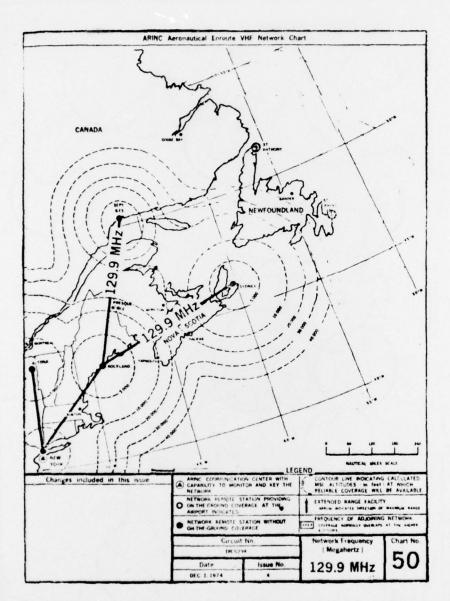


FIGURE C-6. EXTENDED RANGE VHF TRANSMITTERS IN THE US AND CANADA, EAST COAST, CANADA.

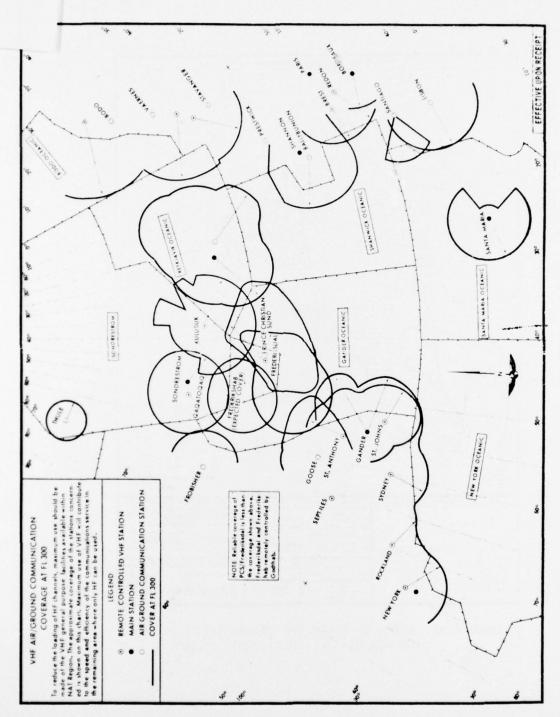


FIGURE C-7. EXTENDED RANGE VHF TRANSMITTERS IN THE NORTH ATLANTIC.

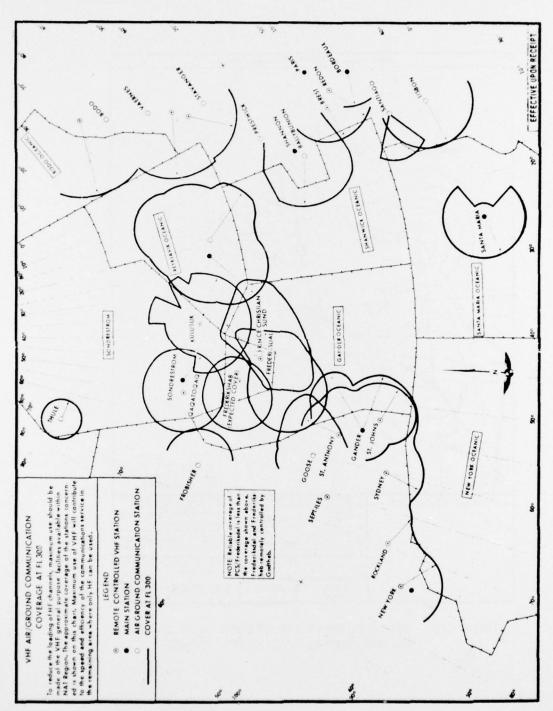


FIGURE C-7. EXTENDED RANGE VHF TRANSMITTERS IN THE NORTH ATLANTIC.

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